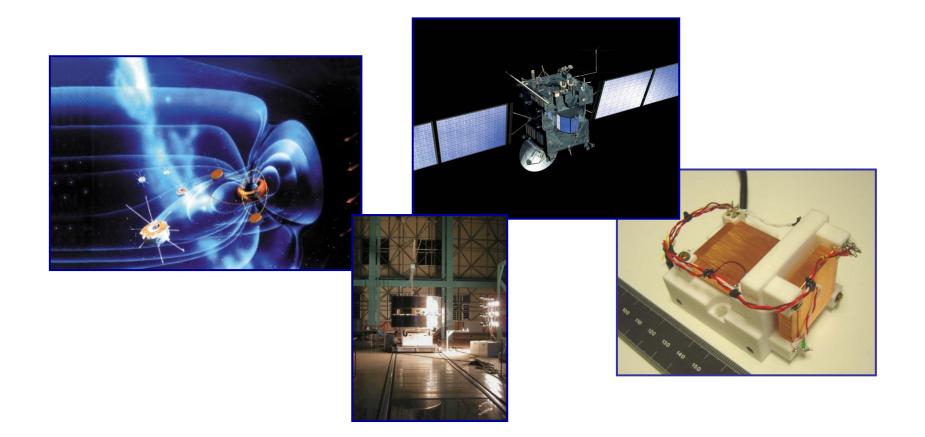
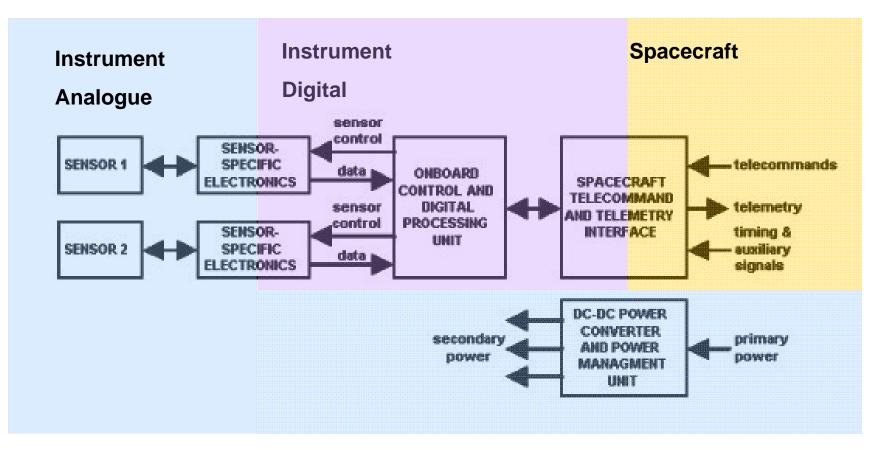
# Instrumentation II Magnetometers and Calibration



### **Generic Space Instrument**



# What do we mean by DC space magnetometer?

- Three B field components in range 0 30Hz
- Wide measurement range 0.01nT 50,000nT
- Robust, reliable, high performance (low noise stable offsets)
- Optimised for power, mass, radiation etc.
- Sensors fitted to a boom away from S/C magnetic disturbance

Sensor Technology	Range (T)	Suitable for space	
SQUID	10 <sup>-14</sup> – 10	No – Cryostat needed	
Optically Pumped	10 <sup>-14</sup> – 10 <sup>-4</sup>	Yes – <b>B</b> and   <b>B</b>	
Fluxgate	10 <sup>-10</sup> – 10 <sup>-4</sup>	Yes – B	
Nuclear Precession	10 <sup>-11</sup> – 10 <sup>-2</sup>	Yes -   <b>B</b>	
Hall Effect	10 <sup>-3</sup> – 10 <sup>-2</sup>	No	
Search Coil	10 <sup>-12</sup> – 10 <sup>6</sup>	Yes for <b>AC</b> fields	

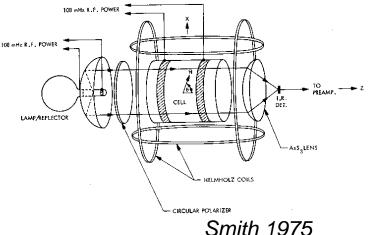




# **Optically Pumped Magnetometers**

- Heritage as a vector magnetometer
- Vector and Scalar Operation (on Cassini)
- Vector Mode
  - RF discharge maintained in a He lamp 1.08um
  - Creates radiation channelled into a He absorption cell
  - He cell atoms are in meta-stable state also by RF discharge
  - Presence of ambient field causes Zeeman splitting
  - Emergent radiation is measured by IR detector
  - The measured absorption depends on efficiency of the optical pumping
  - Helmholtz coils around cell apply rotating sweep fields
  - Signal is obtained by measuring the modulation of rotating sweep fields applied by surrounding Helmholtz coils
  - Results in a sinusoid whose magnitude and phase give the size and direction of the field
  - Signal detected and fedback into the sensor coils





### Scalar mode

- 1.08um radiation and frequency modulated AC field applied.
- Absorption greatest when AC frequency = Larmor frequency.
- Larmor frequency related to |B| by fundamental constants
- Result is a very accurate measure of absolute field

# **Proton Precession Magnetometers**

- Proton rich material eg distilled water .
- Surrounded by induction coil
- AC field induces proton precession •
- Once induced field switched off
- Protons relax back to ambient field precession
- This induces a small AC signal in coil •
- Proportional to ambient field •
- Suitable for slow varying fields
- Used for absolute measurement of **B**
- Used on Earth mapping missions eg Oested, CHAMP

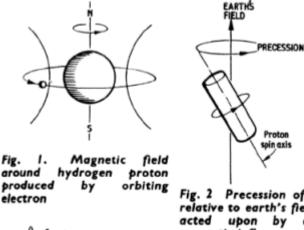


Fig. 2 Precession of proton relative to earth's field when acted upon by external magnetic influence



Fig. 3. Diminishing alternating voltage set up by precession frequencies from the detector coils

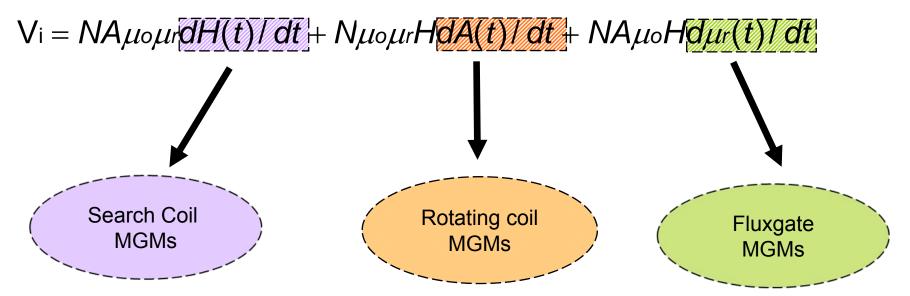
Huggard 1970



# **Induction Magnetometers**

Faraday induction law  $\rightarrow V_i = d\Phi / dt$ ... = d(BA) / dt Since  $B = \mu_0 \mu r H$  $V_i = d(NA\mu_0\mu r(t)H(t) / dt)$ 





# **Anatomy of a Fluxgate**

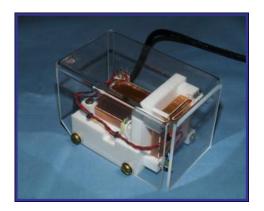
- Operating Principle
  - Soft permeable core driven around hysterisis loop
  - H<sub>EXT</sub> results in a net changing flux
  - Field proportional voltage induced in sense winding
  - Closed loop improves linearity

### Advantages

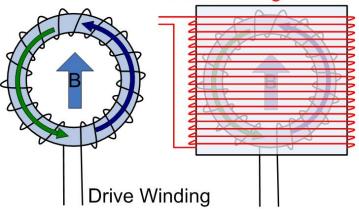
- Low noise ~ 20pT/  $\sqrt{Hz}$  @1Hz
- Wide dynamic range
- Mature technology
- Relatively inexpensive

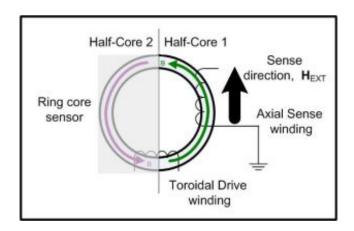
### Disadvantages

- Sensor mass
- Sensor offset
- Power ~ 1W
- In-flight calibration overhead



Sense Winding





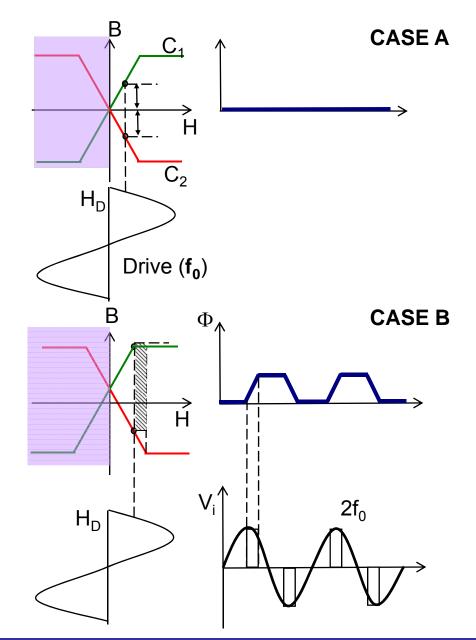
### **CASE A: Zero external DC field**

Half cores saturate synchronously – no net change of flux **seen** by sense wining

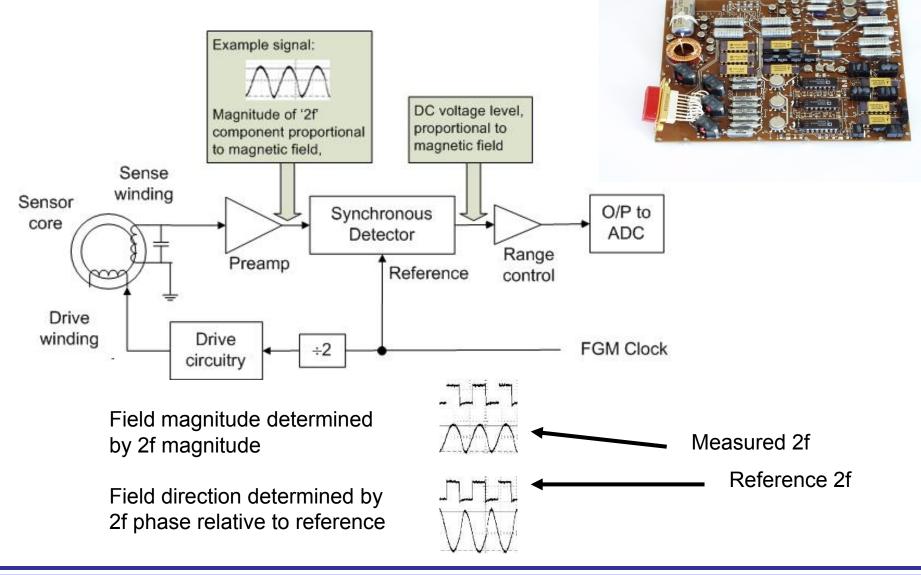
### **CASE B: Non-zero external DC field**

Half cores do not saturate synchronously – a net change of flux **seen** by sense wining

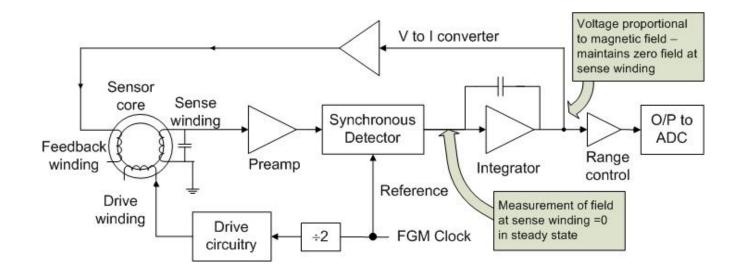
**Change** of flux in sense winding at the 4 crossing of the B-H infection points in each drive period  $\rightarrow$  induced voltage at 2 x f<sub>o</sub> according to Faraday



### Fluxgate Control Electronics: Open Loop



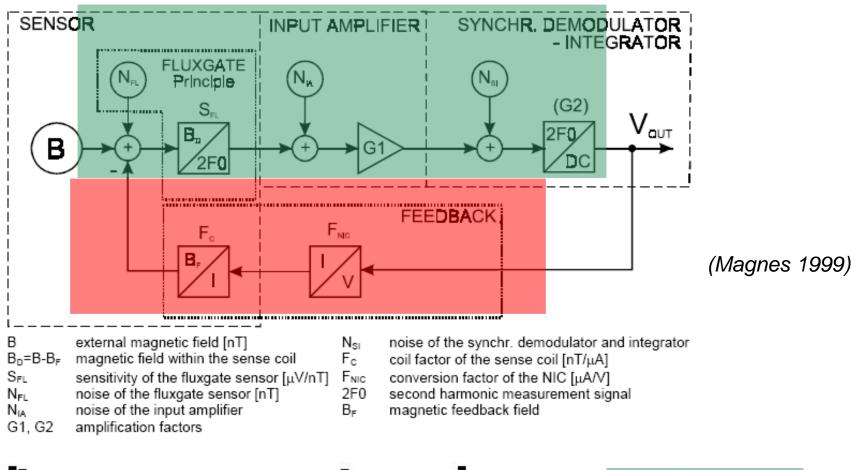
### Fluxgate Control Electronics: Closed Loop



Benefits include improved linearity and temperature stability. Scale factor depends only on feedback resistor/gain stage and coil constant.

Considerable effort spent minimising even harmonics in drive signal some odd harmonics due to transformer effect.

Includes anti-aliasing filter



$$\begin{bmatrix} [(B+N_{FL}-B_F)\cdot S_{FL}]+N_{LA}]\cdot G_1+N_{SI} \end{bmatrix} \cdot G_2 = V_{OUT}$$
 Measured signal  
$$B_F = F_{NIC} F_C V_{OUT} = k V_{OUT}$$
 Feedback signal

Equating terms and re-arranging

$$V_{OUT} = B \cdot \frac{S_{FL}G_2G_1}{1 + k S_{FL}G_2G_1} + \frac{(N_{FL}S_{FL} + N_{LA}) \cdot G_2G_1 + N_{SI}G_2}{1 + k S_{FL}G_2G_1}$$

And if  $kS_{FL}G_2G_1 >> 1$ 

$$V_{OUT} = B \cdot \frac{1}{k} + \frac{N_{FL}}{k} + \frac{N_{LA}}{k S_{FL}} + \frac{N_{SI}}{k S_{FL} G_1}$$

### **Two conclusions**

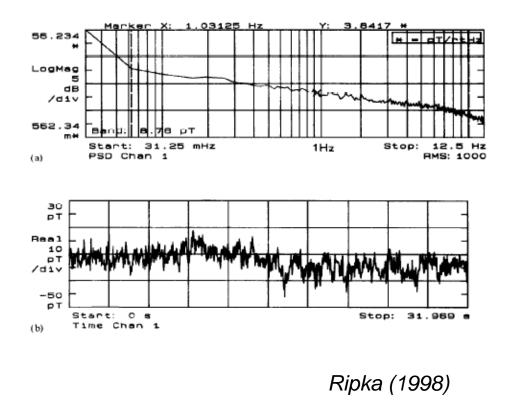
Measurement range only set by feedback circuit

Output noise is dominated by input amplifier and sensor noise only

(Very low noise analogue pre-amps available)

# **Fluxgate Noise**

- Best expressed as a Noise Spectral Density (NSD) often at1Hz
- Characteristic typically has a 1/f fall off



 Between 0 and Nyquist can use following expression to calculate RMS Noise

$$N_{\rm rms} = \left( \int_{f_{\rm L}}^{f_{\rm H}} P(f) \, dt \right)^{1/2} = (P(1) \ln(f_{\rm H}/f_{\rm L}))^{1/2}$$

- Above Nyquist noise will be flat (ie white noise) due to ADC quantization
- Best quality fluxgates have NSD ~5pT/Root Hz at 1Hz

# Imperial fluxgate instrument performance

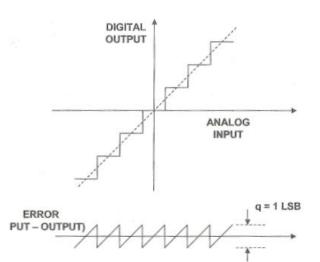
- Industrial partner Ultra Electronics
- Cassini/Double Star Heritage
- Two core sensor
- Tuned second harmonic detection
- Dual sense and feedback windings
- Offset stability < 0.05 nT/°C</li>
- Scale factor drift < 40 ppm/°C</li>
- Noise density < 8pT/root Hz @1Hz</li>
- Operating range
  - -80°C to 70°C (operational)
  - -130°C to 90°C (non-operational)

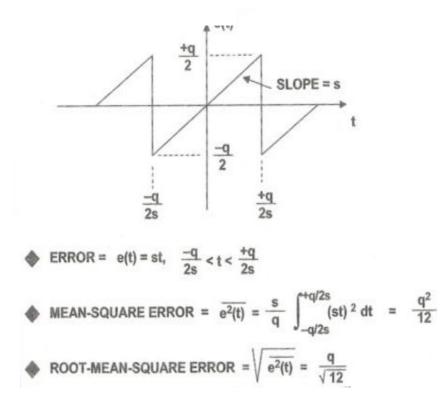




# **Importance of ADC: Quantization Noise**

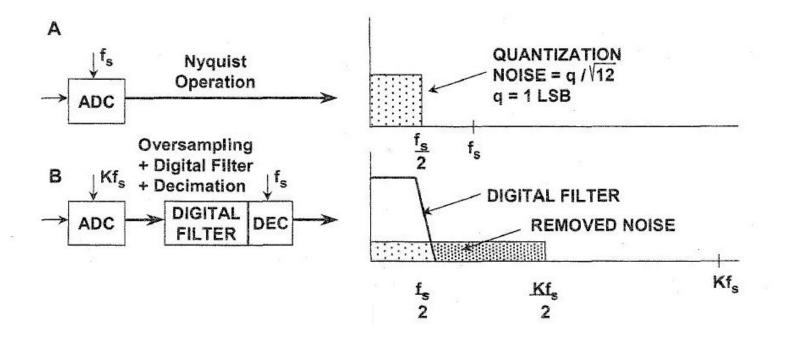
- Large number of bits N
- Ideal linearity
- No missing codes
- Radiation tolerance
- Ideal quantization noise





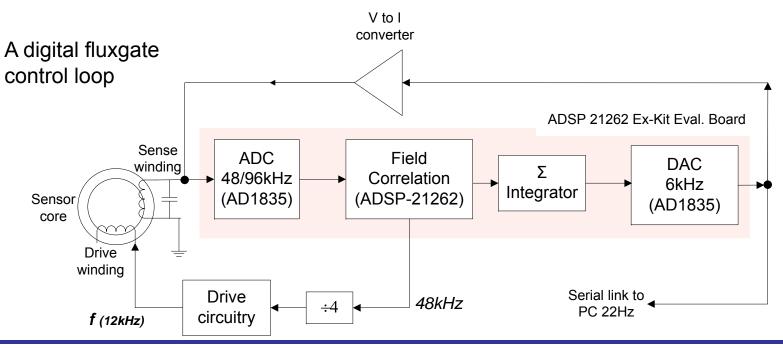
# **Quantization Noise**

- Large N Rad-tolerant ADCs are a 'big' problem for all instruments
- Solution: MIL-STD devices with spot shields (N ~14)
- Traditionally a separate self contained card Cluster, Rosetta, Cassini
- Use oversampling to reduce Q noise
- Q noise should be matched to intrinsic sensor noise based on desired range, resolution sensor scale factor and N and LSB

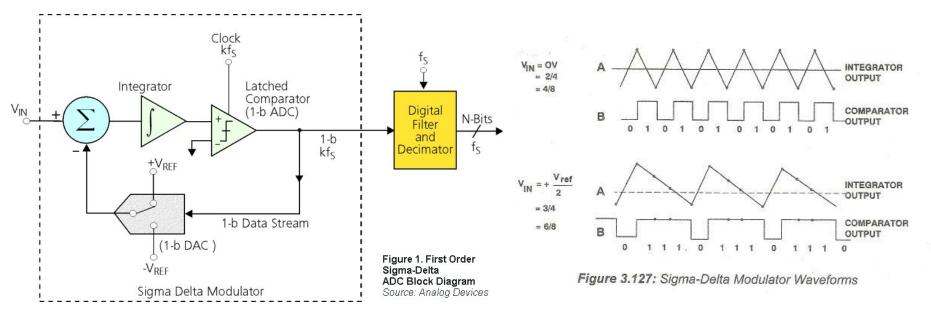


# **Digital Magnetometers**

- Means migrating control loop into digital domain
- ADC and DAC utilised within sensor control loop
- Offers increased flexibility programmable
- First Missions late 90s ROMAP, VEX, Astrid, Oersted
- Shown to reduce analogue content and power consumption
- Numerous designs still being played out a very active field

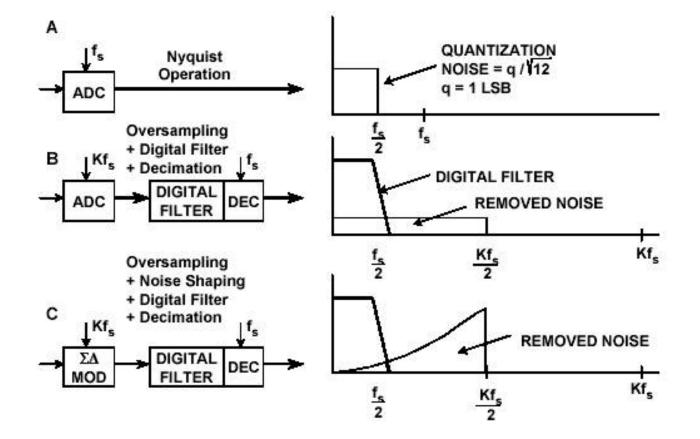


# Delta Sigma Fluxgates - A hot topic

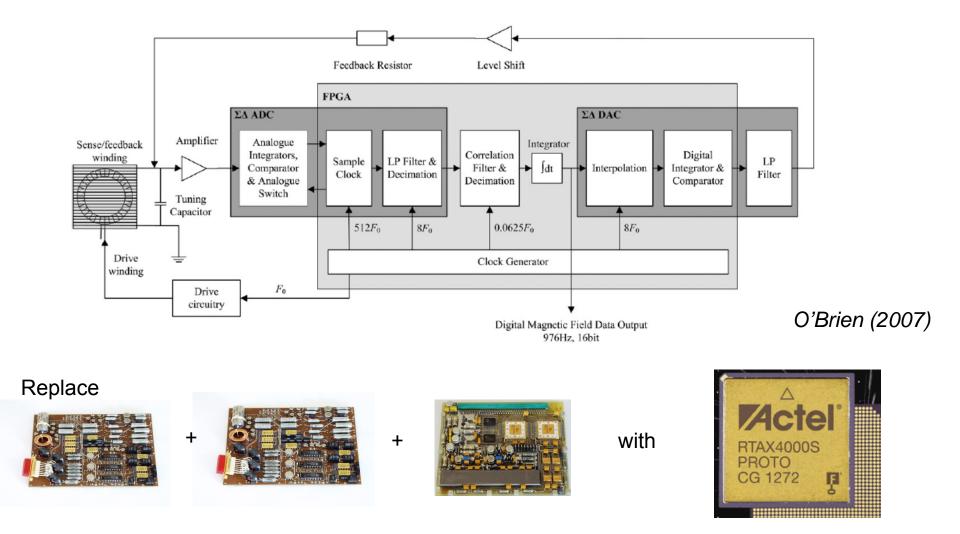


- Single bit quantization at very high frequency- linear by definition
- Tracks changes in consecutive samples rather than absolute value
- 'Ones' density of the 1 bit data stream provides an average value of Vin
- Can be implemented with a rad-hard analogue discretes and rad-hard digital logic mixed signal ASIC
- Additional gain due to noise shaping
- Eliminates need for old fashioned non rad-hard ADCs

# Noise shaping effect



# **Delta-Sigma Magnetometer**



# **Anisotropic Magnetoresistance**

### Magneto Resistance Effect

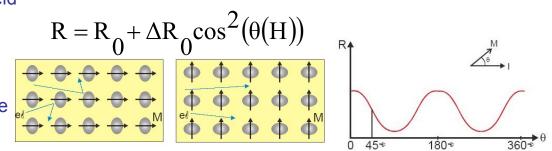
- Change of resistance in magnetic field
- AMR single layer permalloy,
- AMR  $\Delta$ R/Rmin of order 1- 2%
- AMR has lowest noise floor
- Johnson noise limited no shot noise

### Barber Poles

- Max, sensitivity & linearity at M v H 45°
- Conductive strips for linear operation

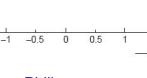
### AMR Sensors

- Thin film solid state devices
- Implemented as Wheatstone bridge
- Mass <1g, Ceramic package</li>
- Sensitivity increases with increasing bridge voltage, V<sub>B</sub>



0

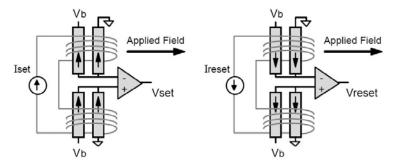
# $y \stackrel{1}{\leftarrow} \stackrel{\text{Permalloy}}{\leftarrow} x \qquad Barber pole \\ \downarrow \stackrel{1}{\leftarrow} \stackrel{1$

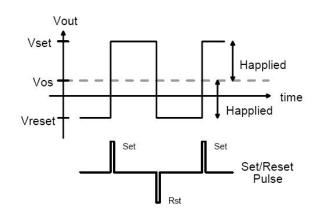


Imperial College London MBH615

# **Integrated coils**

- Set Reset Coils
  - Planar coils around each bridge resistor
  - Coil axis parallel to Easy axis
  - Used to re-align the anisotropic direction
  - Large current spike needed
  - Can extract sensor offset (unlike fluxgate)
  - Requires de-modulation to DC
  - Compensates for offset and offset drift
  - Improves sensor noise floor

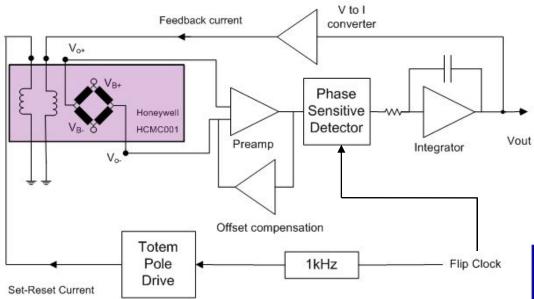




### Offset coils

- Integrated coils around the bridge
- Coil axis parallel to Hard (sensitive) axis
- Permits electromagnetic feedback
- Used in closed loop back off measured field
- Improves linearity and variation of sensitivity with temperature
- Suppresses Barkhausen noise

# Single axis AMR magnetometer

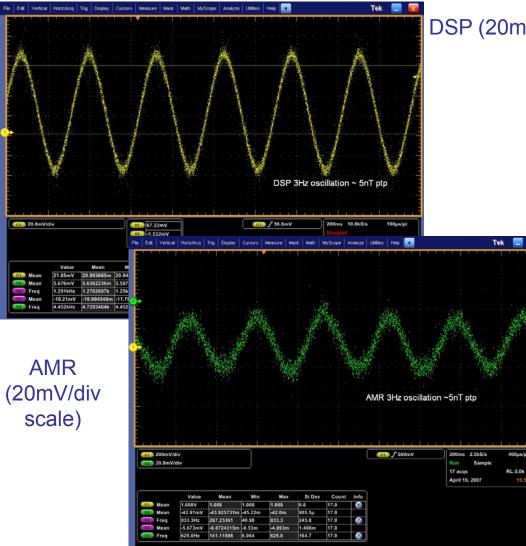


$$V_{o} = H_{y} \times \frac{R_{FB}}{A_{COIL}}$$

Analog build Set-Rest 4A with 2µs  $\tau_{C}$  Sensitivity proportional to  $V_{B}$  Closed loop



# Stimulus measurement – Fluxgate vs AMR



DSP (20mV/div scale)

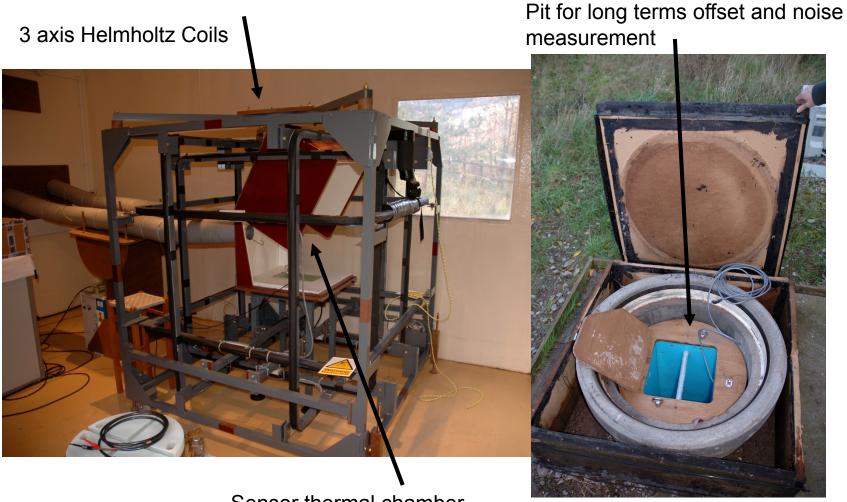
- Three layer Mu-Metal shield
- 3Hz sine wave 5nT ptp
- Optimal AMR configuration
- Closed loop, R<sub>FB</sub>=9kΩ
- Bridge voltage 12V
- Offset compensation
- Flip frequency, 1.1kHz
- Sensitivity ~ 11mV/nT
- Sensitivity not linear with increasing R<sub>FB</sub>
- Some residual offset in closed lop
- Temperature measurement
  outstanding

# **Calibration equation for a vector magnetometer**

- **Calibration Matrix** 12 paramaters needed to transform measured volts to accurate field components into a physically useful co-ordinate system eg GSE, GSM
  - Calibration Matrix
    - Sensor gains convert from raw volts to nT
    - Sensor mis-alignments correct from deviation from nominal sensor axis
    - Euler **angles** –transform othogonalised components into required system
  - Offset vector:
    - Sensor offset correct for zero level readings (due to sensor, electronics or S/C)
  - Calibration Files
    - Text files with calibration matrix & offset vector for each sensor on a daily or orbit basis :

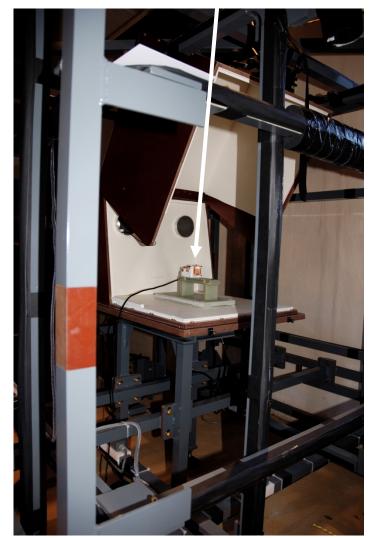
$$\begin{pmatrix} B_x \\ B_y \\ B_z \end{pmatrix} = \begin{pmatrix} c_{11} & c_{12} & c_{13} \\ c_{21} & c_{22} & c_{23} \\ c_{31} & c_{32} & c_{33} \end{pmatrix} \begin{pmatrix} B_{S1} - O_1 \\ B_{S2} - O_2 \\ B_{S3} - O_3 \end{pmatrix}$$

# **Imperial's Magnetic Coil Facility**



Sensor thermal chamber

### Sensor under test





 Facility dynamically backs of Earth's field using
 Two Earth Field Reference Magnetometers (EFR) located either side of the hut

•EFR located in pits either side of test hut •Sum (average) of EFRs used to cancel Earths field inside coil system

•Difference (gradient) of EFRs used for monitoring

# **Practical calibration models**

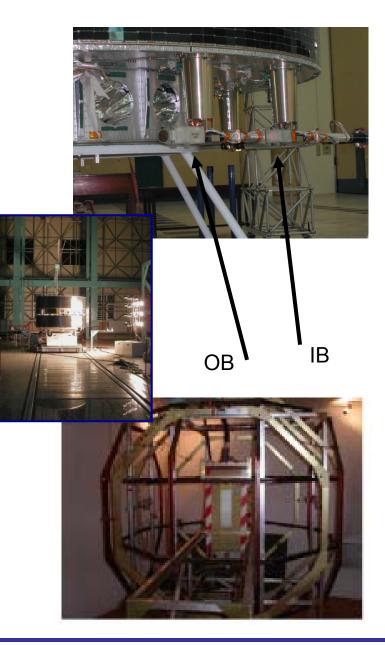
- Ground Calibration we determine
  - Sensor calibration parameters on ground,
  - Their associated temperature coefficients,
  - Their variations with input power
  - The sensor noise

### Magnetic Cleanliness Program - includes

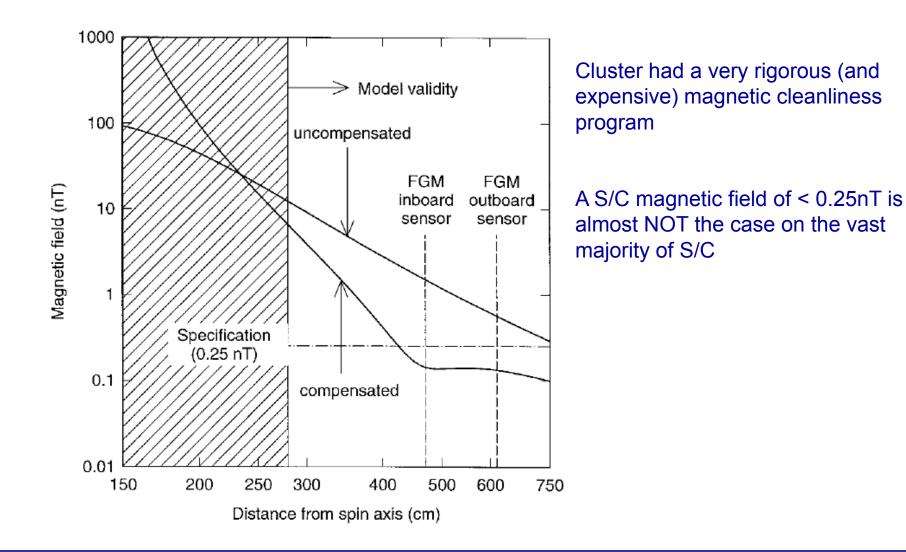
- Maximum length boom
- Low field requirement at boom tip
- Magnetic screening of materials and units
- A spacecraft magnetic model
- System level magnetic test

### In-flight

- range switching, calibration steps
- In-flight calibration techniques
- Use of multiple sensors
- Use of absolute and vector sensors
- Use of dual-gradiometer modes

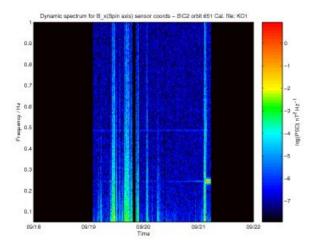


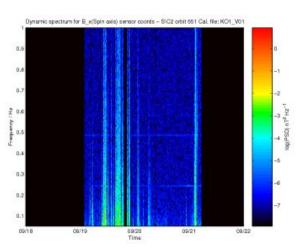
# System Level Magnetic Test: Cluster Example



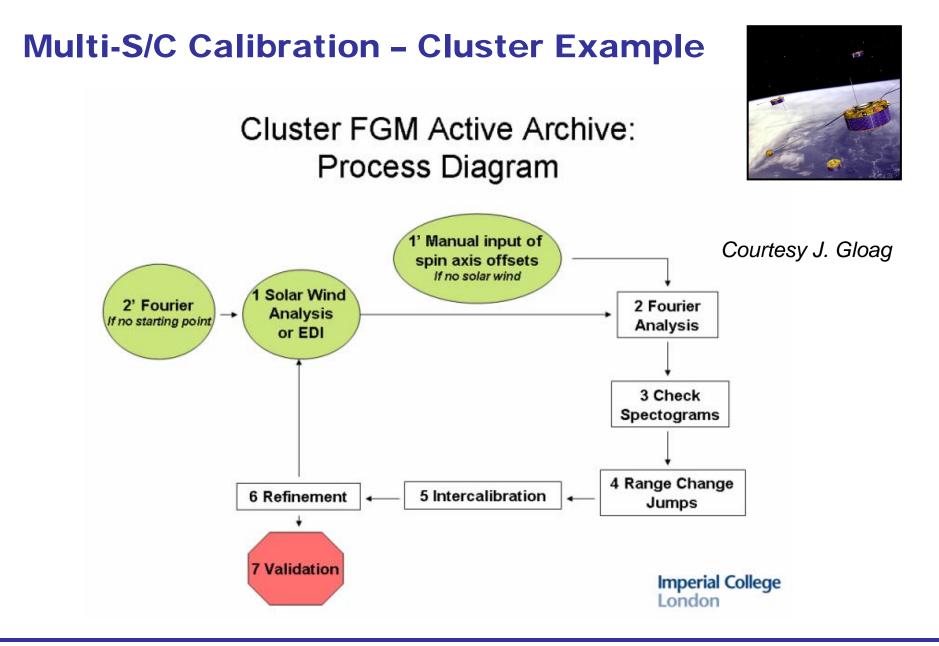
# In-flight calibration techniques

- Spin stabalised spacecraft
  - Fourier analysis on spinning data
  - Permits recovery of 8 of the 12 cal parameters
  - Major error spin axis offset
  - Residual spin tone indicates calibration error
  - Example Missions:
    - Cluster, Ulysses, Double Star, Equator-S, Themis
- Three-axis stabalised spacecraft
  - More difficult to calibrate
  - Utilise S/C rolls for offset measurement
  - Statistical analyses of solar wind data
  - Looks for correlations between **B** and B components
  - Additional absolute reference magnetometer useful
  - Example Missions
    - Cassini, Rosetta, Oersted, Venus Express



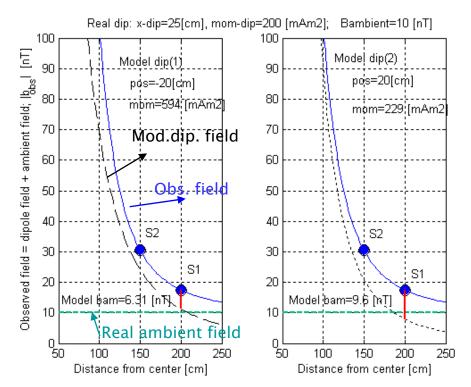


• Multiple spacecraft missions = multiple calibration references



# **Dual Magnetometer Mode**

- Used in cases where S/C field contaminates measurement
- IB and OB sensor used as a gradiometer
- Ambient field same at both IB & OB
- S/C field NOT same at IB & OB
- Number of sensors is proportional to multipole moment that may be extracted
- Two sensors limit model to a dipole of fixed position
- Other techniques utilising pattern recognition in operation
- Relative sampling of both sensors important especially on spinning S/C
- Usually results in reduced data rates
- Example missions: Double Star, Venus Express

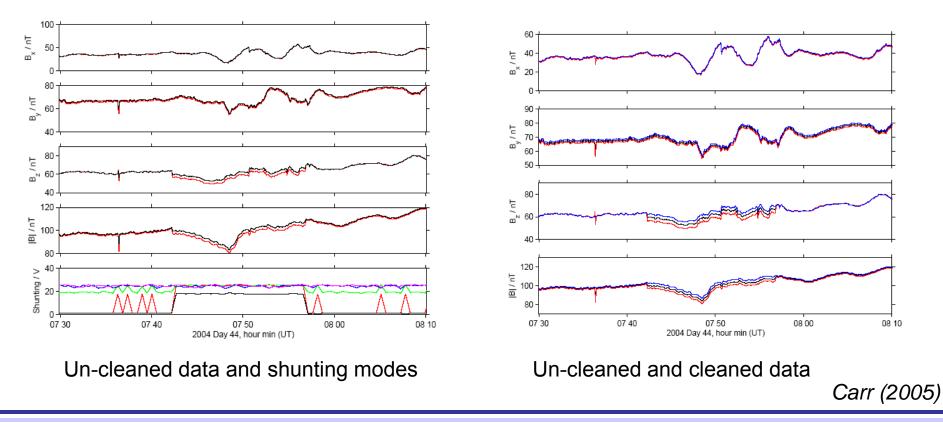


### Figure courtesy Delva

Example (1 dim.): Real Bam=10 nT Real SC dipole: p =25 cm m =200 mAm2	Solution 1: p~ = - 20cm m~=594mA <sup>2</sup> Bam~=6.31nT	Solution 2: p~ =+20cm m~=229mAm <sup>2</sup> Bam~=9.6nT
m =200 mAm2	Bam~=6.3111	Bam~=9.601

# **Case Study. Double Star magnetometer**

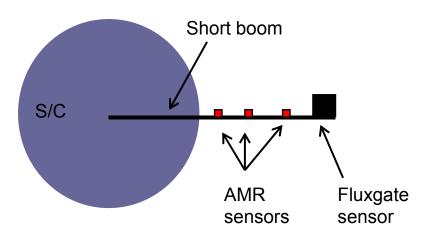
- OB sensor 5m, IB sensor 3.5m
- Spin synchronised disturbance due to unbalanced solar array current
- Amplitude varies with S/C shunting mode
- Data cleaned using gradiometer mode
- Resulting data set is spin averaged resolution (0.25Hz) compared to 11Hz on-board



# A new magnetometer model?

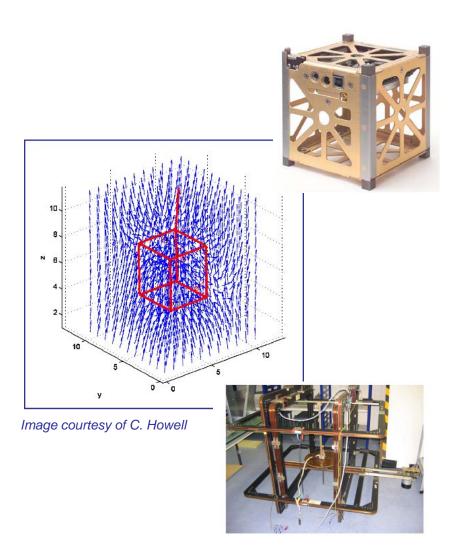
- Fluxgate AMR combination
- Single fluxgate at end of a (shorter boom)
- Several AMR sensors inward of the fluxgate
- Permits multipole expansion of S/C field
- Accurate separation ambient field at instrument intrinsic data rate
- Precise tracking of fluxgate offsets
- Required for space plasma constellations
- Potential for automation
- Could be applied acoss missions
- Extendable to an array of AMR sensors

Question – How to validate concept ?



# A magnetometer array

- Imperial College student satellite program
  - Milestone Two spacecraft in LEO
  - 10cm cube, 1kg modules
  - Injection into LEO approx \$30,000
- Aims
  - Measure ULF wave field in dayside magnetosphere
  - Flight qualify FPGA controlled AMR array
  - Validate S/C field rejection algorithms
  - Extract accurate magnetic field vector
- Ground validation
  - Mobile Coil Facility
  - ESTEC MDM to calculate E-box moments
  - Measure both S/C components and assembled S/C
  - Measured moments fitted to S/C model
  - Permits validation test of field rejection on ground



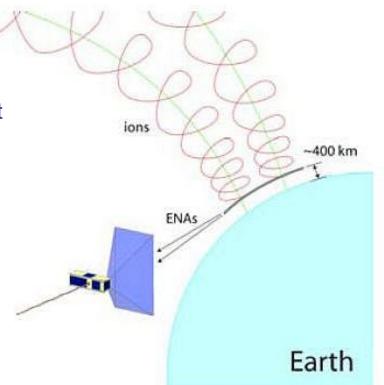


# **Potential Flight Opportunity 2010**

# CINEMA- CubeSat for measurement of ions, neutrals and magnetic fields

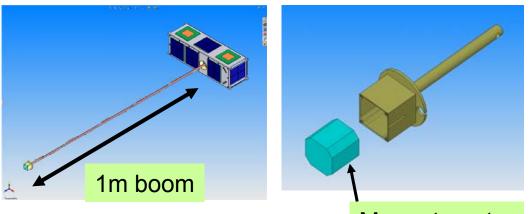


- Space plasma science measurement on 3U CubeSat
- Led by UCB/SSL
- LEO with >65° inclination (72° nominal), 650km
- 1m deployable boom
- Spin stabalised at ~1rpm
- Two MAG sensors
- Submitted to NSF Space Weather Competition 2008
- To be re-submitted 2009
- Heritage: GeneSat & STEREO





# **CINEMA Magnetometer**



Magnetometer

1m extendable boom Boom mass ~120g MAG orientation not controlled Determined by magneto-torquer pulse post deployment Following de-tumble CubeSat spun up and spin axis aligned normal to ecliptic

MAGIC Magnetometer Modes <u>Attitude Mode</u> Accuracy <25nT, <150mW <u>Science Mode</u> Accuracy <2nT, <750mW Instrument Range +/-65536nT Resolution: 0.25nT



MAGIC Sensor head



MAG Boom Harness