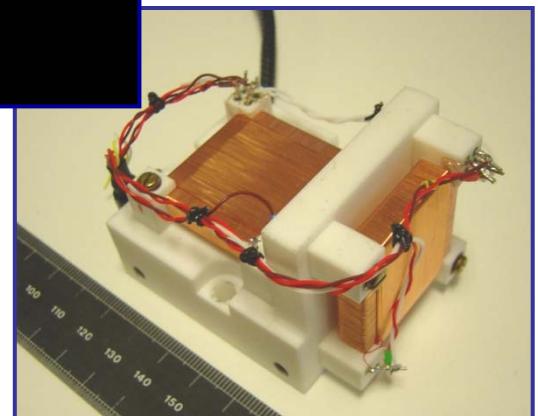
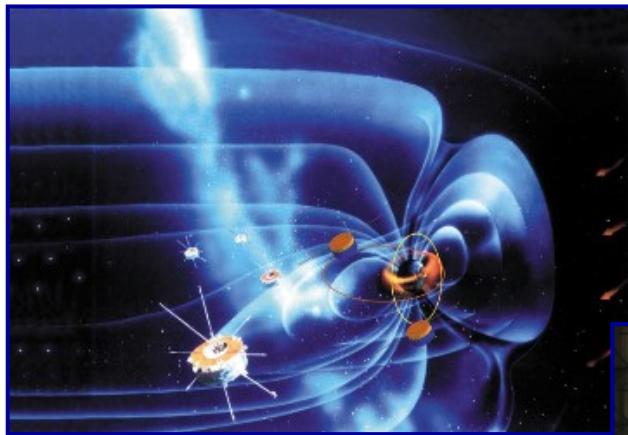
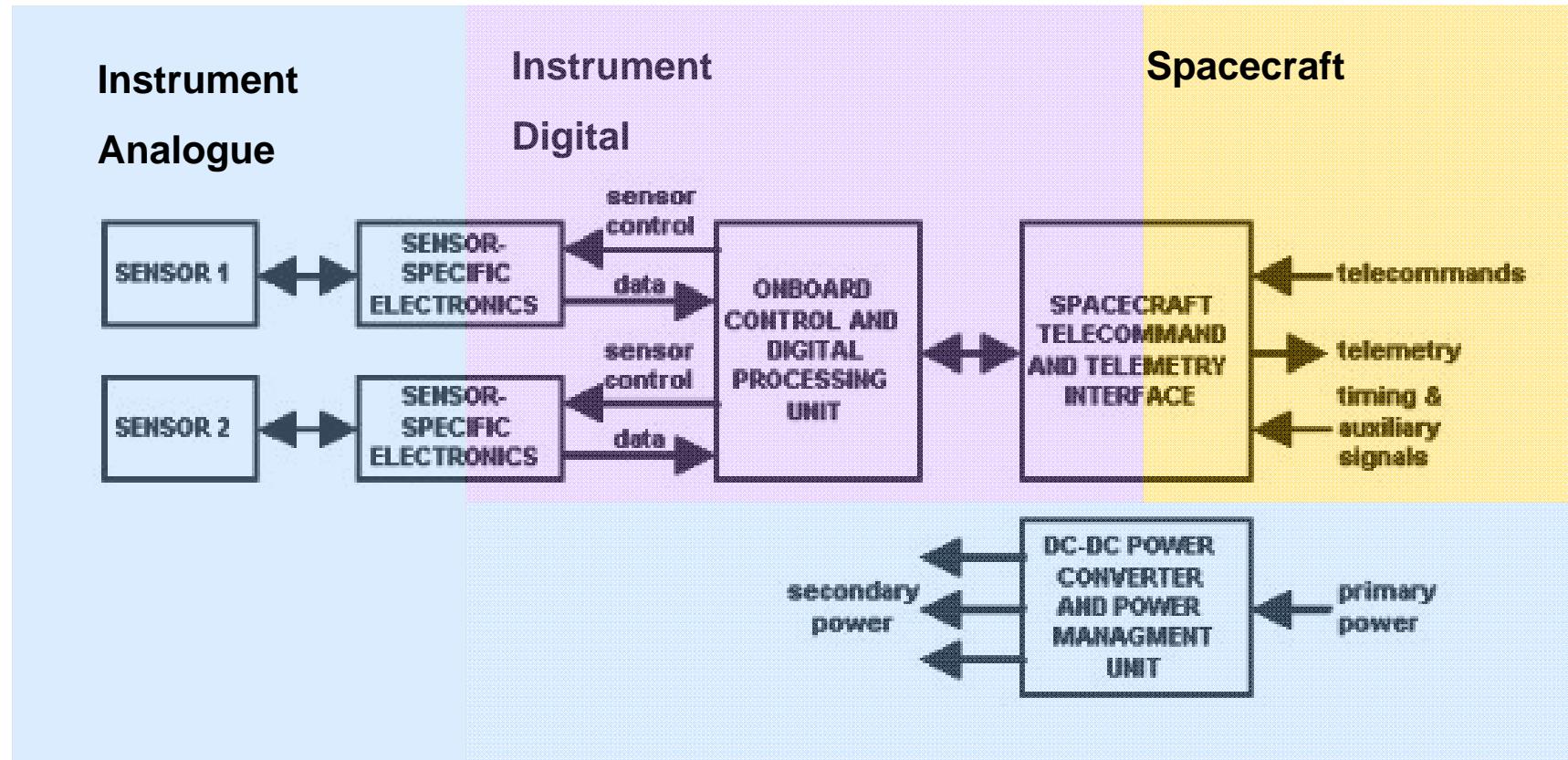


# Instrumentation II

## Magnetometers and Calibration



# Generic Space Instrument



# What do we mean by DC space magnetometer?

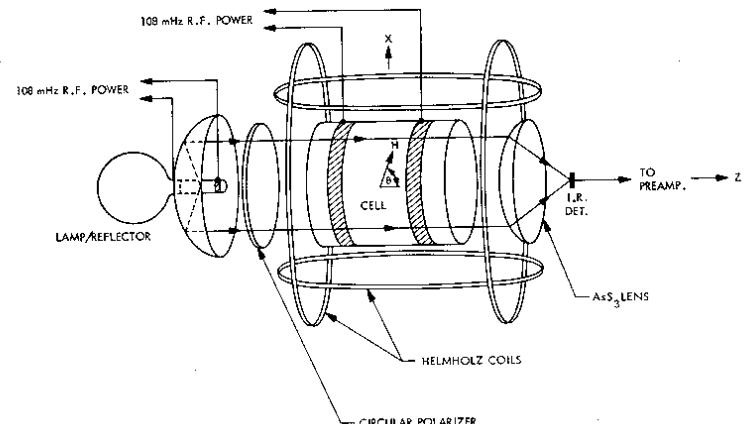
- Three B field components in range 0 - 30Hz
- Wide measurement range 0.01nT – 50,000nT
- Robust, reliable, high performance (low noise – stable offsets)
- Optimised for power, mass, radiation etc.
- Sensors fitted to a boom away from S/C magnetic disturbance

Sensor Technology	Range (T)	Suitable for space
SQUID	$10^{-14} - 10$	No – Cryostat needed
Optically Pumped	$10^{-14} - 10^{-4}$	Yes – <b>B</b> and $ \mathbf{B} $
Fluxgate	$10^{-10} - 10^{-4}$	Yes – <b>B</b>
Nuclear Precession	$10^{-11} - 10^{-2}$	Yes - $ \mathbf{B} $
Hall Effect	$10^{-3} - 10^{-2}$	No
Search Coil	$10^{-12} - 10^6$	Yes for <b>AC</b> fields



# Optically Pumped Magnetometers

- **Heritage as a vector magnetometer**
- **Vector and Scalar Operation (on Cassini)**
- **Vector Mode**
  - RF discharge maintained in a He lamp – 1.08um
  - Creates radiation - channelled into a He absorption cell
  - He cell atoms are in meta-stable state also by RF discharge
  - Presence of ambient field causes Zeeman splitting
  - Emergent radiation is measured by IR detector
  - The measured absorption depends on efficiency of the optical pumping
  - Helmholtz coils around cell apply rotating sweep fields
  - Signal is obtained by measuring the modulation of rotating sweep fields applied by surrounding Helmholtz coils
  - Results in a sinusoid whose magnitude and phase give the size and direction of the field
  - Signal detected and feedback into the sensor coils
- **Scalar mode**
  - 1.08um radiation and frequency modulated AC field applied.
  - Absorption greatest when AC frequency = Larmor frequency.
  - Larmor frequency related to  $|\mathbf{B}|$  by fundamental constants
  - Result is a very accurate measure of absolute field



Smith 1975

# Proton Precession Magnetometers

- Proton rich material eg distilled water
- Surrounded by induction coil
- AC field induces proton precession
- Once induced field switched off
- Protons relax back to ambient field precession
- This induces a small AC signal in coil
- Proportional to ambient field
- Suitable for slow varying fields
- Used for absolute measurement of **B**
- Used on Earth mapping missions eg Oested, CHAMP

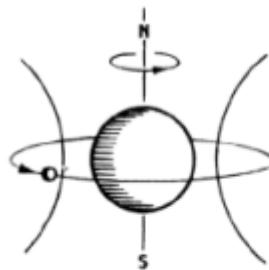


Fig. 1. Magnetic field around hydrogen proton produced by orbiting electron

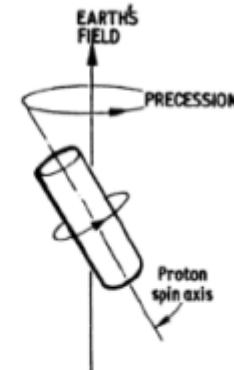


Fig. 2 Precession of proton relative to earth's field when acted upon by external magnetic influence

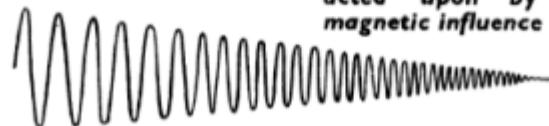


Fig. 3. Diminishing alternating voltage set up by precession frequencies from the detector coils

Huggard 1970



# Induction Magnetometers

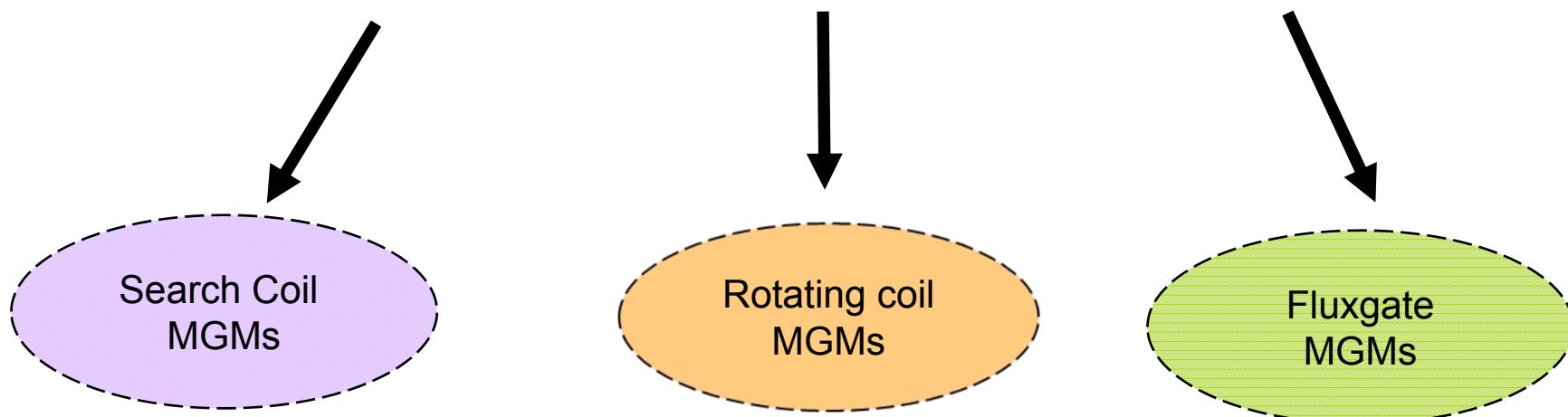
Faraday induction law  $\rightarrow$   $V_i = d\Phi / dt$

$\dots = d(BA) / dt$  Since  $B = \mu_0 \mu_r H$

$$V_i = d(NA\mu_0 \mu_r(t)H(t) / dt)$$

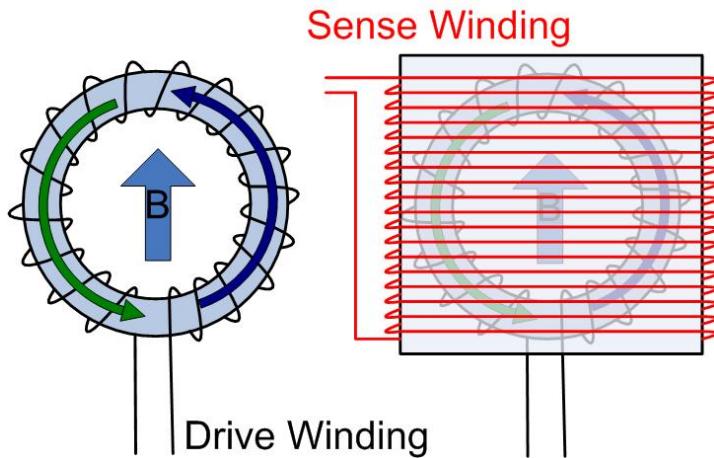
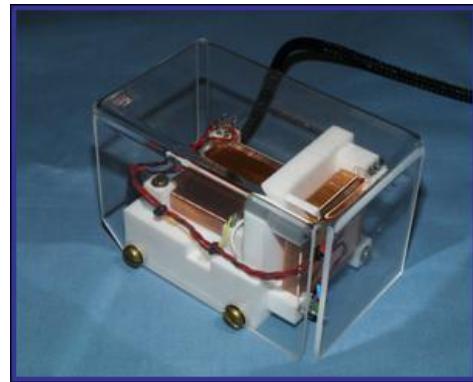
## Expanded

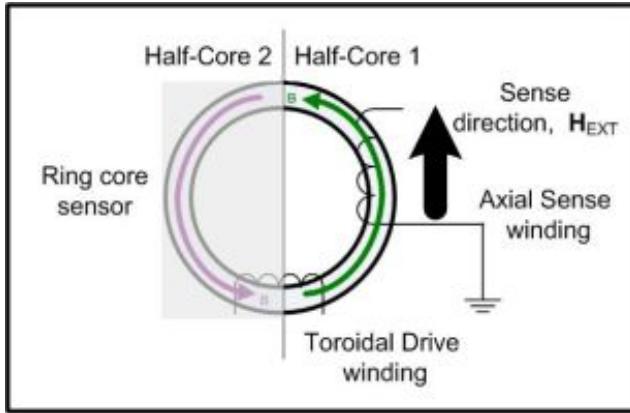
$$V_i = NA\mu_0 \mu_r [dH(t) / dt] + N\mu_0 \mu_r H [dA(t) / dt] + NA\mu_0 H [d\mu_r(t) / dt]$$



# Anatomy of a Fluxgate

- **Operating Principle**
  - Soft permeable core driven around hysteresis loop
  - $H_{EXT}$  results in a net changing flux
  - Field proportional voltage induced in sense winding
  - Closed loop improves linearity
- **Advantages**
  - Low noise  $\sim 20\text{pT}/\sqrt{\text{Hz}}$  @1Hz
  - Wide dynamic range
  - Mature technology
  - Relatively inexpensive
- **Disadvantages**
  - Sensor mass
  - Sensor offset
  - Power  $\sim 1\text{W}$
  - In-flight calibration overhead





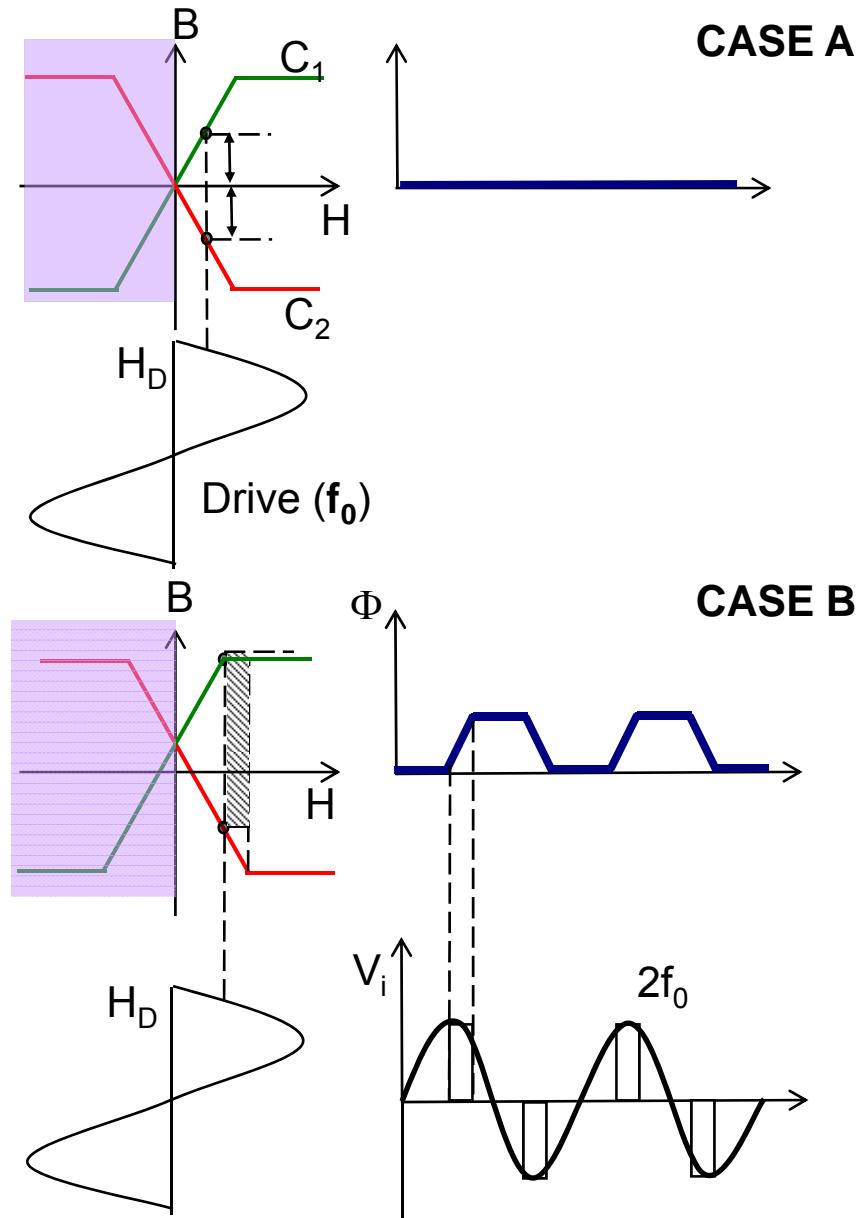
### CASE A: Zero external DC field

Half cores saturate synchronously – no net change of flux **seen** by sense winding

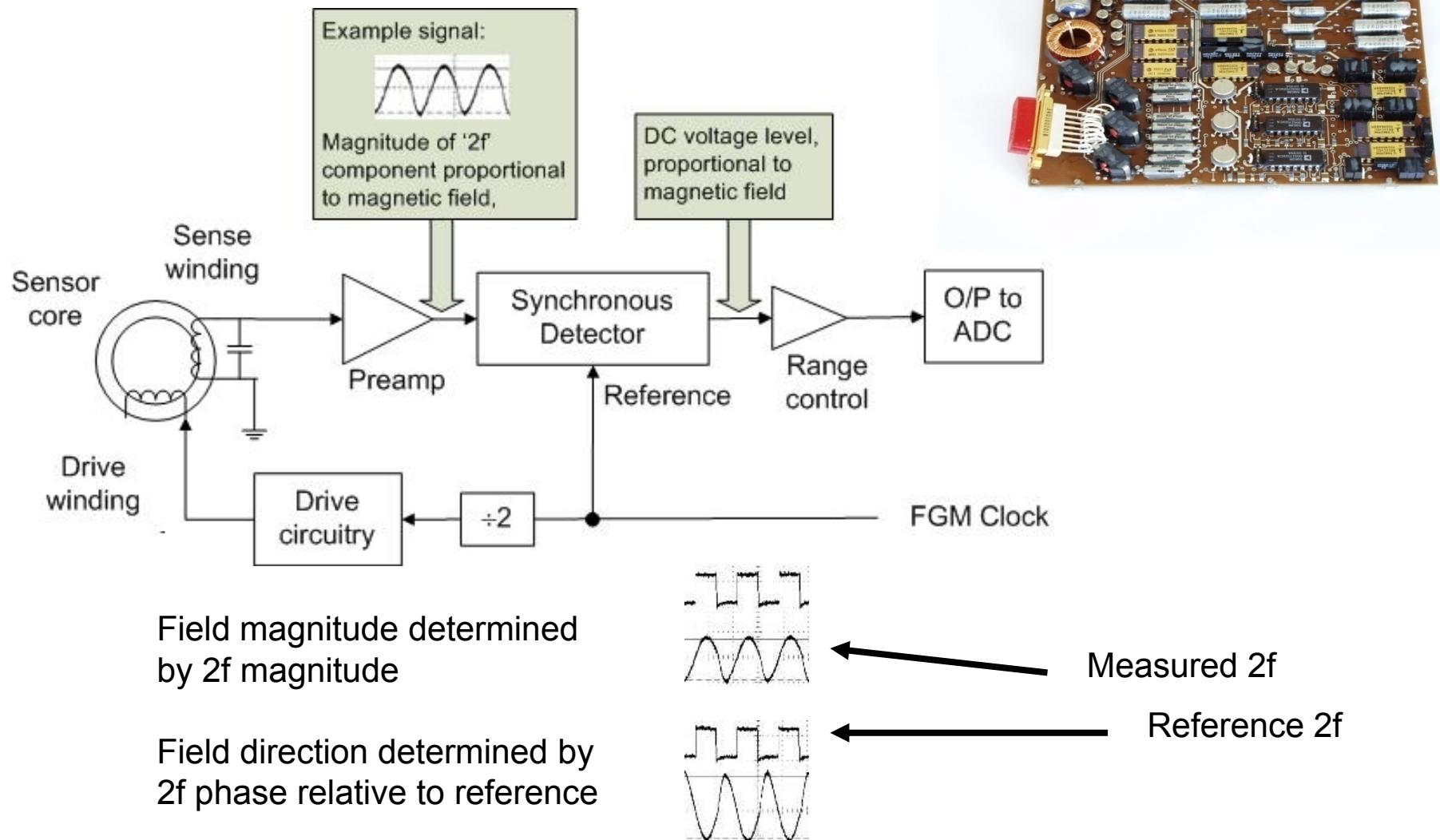
### CASE B: Non-zero external DC field

Half cores do not saturate synchronously – a net change of flux **seen** by sense winding

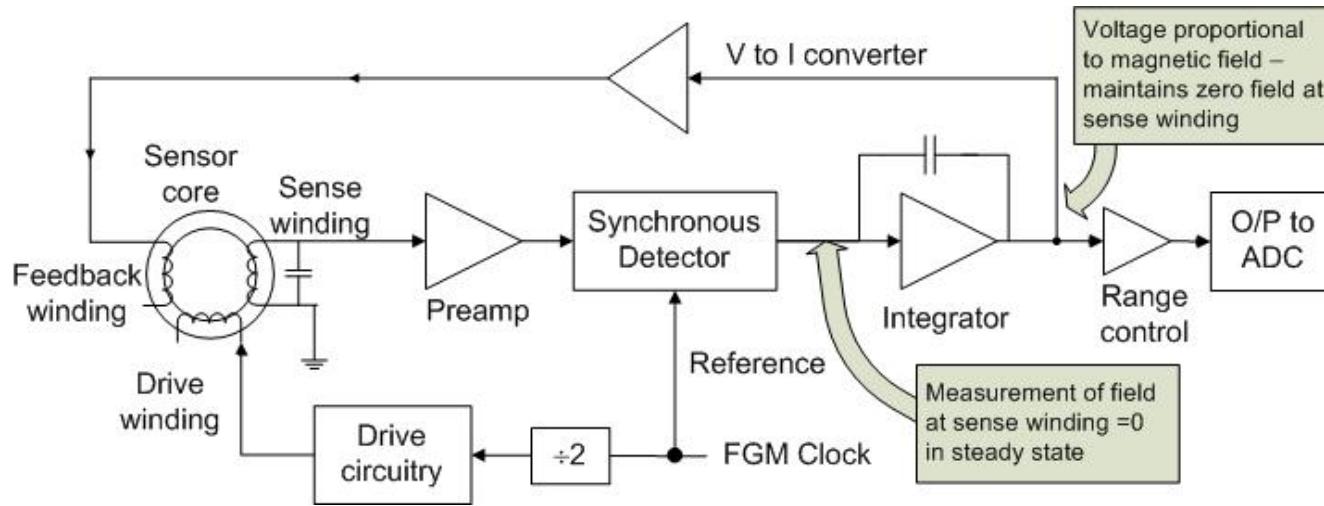
**Change** of flux in sense winding at the 4 crossing of the B-H inflection points in each drive period  $\rightarrow$  induced voltage at  $2 \times f_0$  according to Faraday



# Fluxgate Control Electronics: Open Loop



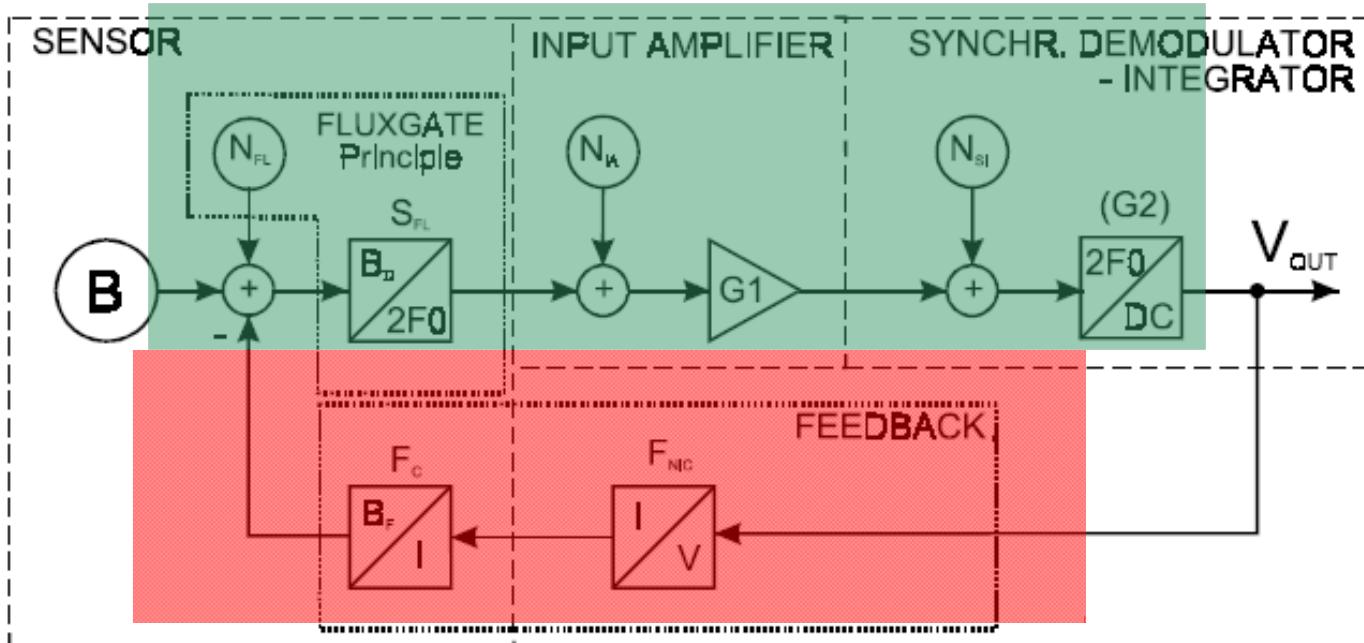
# Fluxgate Control Electronics: Closed Loop



Benefits include improved linearity and temperature stability. Scale factor depends only on feedback resistor/gain stage and coil constant.

Considerable effort spent minimising even harmonics in drive signal  
some odd harmonics due to transformer effect.

Includes anti-aliasing filter



(Magnes 1999)

$B$	external magnetic field [nT]	$N_{SI}$	noise of the synchr. demodulator and integrator
$B_D = B - B_F$	magnetic field within the sense coil	$F_C$	coil factor of the sense coil [ $\mu\text{V}/\text{nT}$ ]
$S_{FL}$	sensitivity of the fluxgate sensor [ $\mu\text{V}/\text{nT}$ ]	$F_{NIC}$	conversion factor of the NIC [ $\mu\text{A}/\text{V}$ ]
$N_{FL}$	noise of the fluxgate sensor [nT]	2F0	second harmonic measurement signal
$N_{IA}$	noise of the input amplifier	$B_F$	magnetic feedback field
$G_1, G_2$	amplification factors		

$$[[[(B + N_{FL} - B_F) \cdot S_{FL}] + N_{IA}] \cdot G_1 + N_{SI}] \cdot G_2 = V_{OUT} \quad \text{Measured signal}$$

$$B_F = F_{NIC} F_C \quad V_{OUT} = k V_{OUT} \quad \text{Feedback signal}$$

Equating terms and re-arranging

$$V_{OUT} = B \cdot \frac{S_{FL} G_2 G_1}{1 + k S_{FL} G_2 G_1} + \frac{(N_{FL} S_{FL} + N_{IA}) \cdot G_2 G_1 + N_{SI} G_2}{1 + k S_{FL} G_2 G_1}$$

And if  $k S_{FL} G_2 G_1 \gg 1$

$$V_{OUT} = B \cdot \frac{1}{k} + \frac{N_{FL}}{k} + \frac{N_{IA}}{k S_{FL}} + \frac{N_{SI}}{k S_{FL} G_1}$$

## Two conclusions

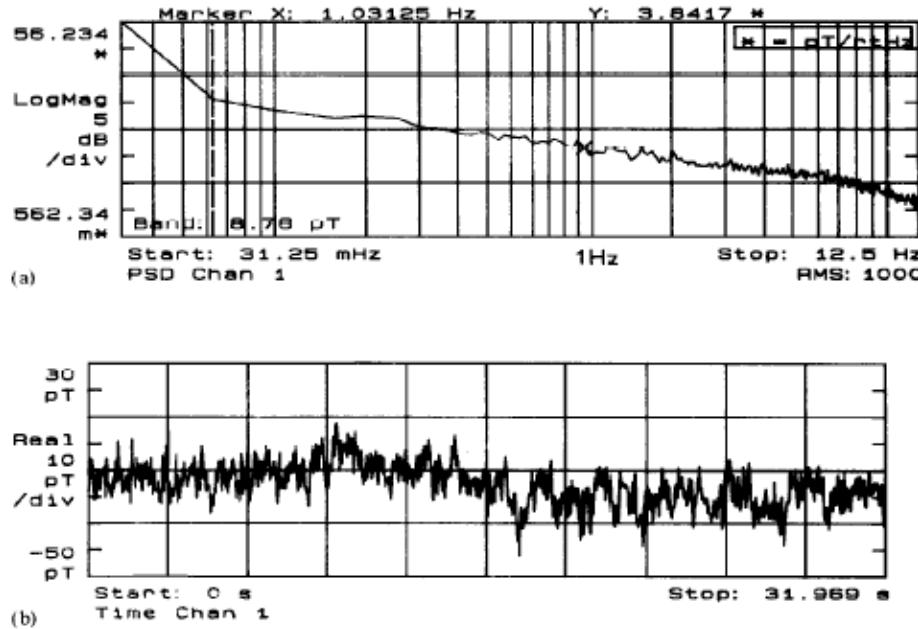
Measurement range only set by feedback circuit

Output noise is dominated by input amplifier and sensor noise only

(Very low noise analogue pre-amps available)

# Fluxgate Noise

- Best expressed as a Noise Spectral Density (NSD) often at 1Hz
- Characteristic typically has a 1/f fall off



- Between 0 and Nyquist can use following expression to calculate RMS Noise

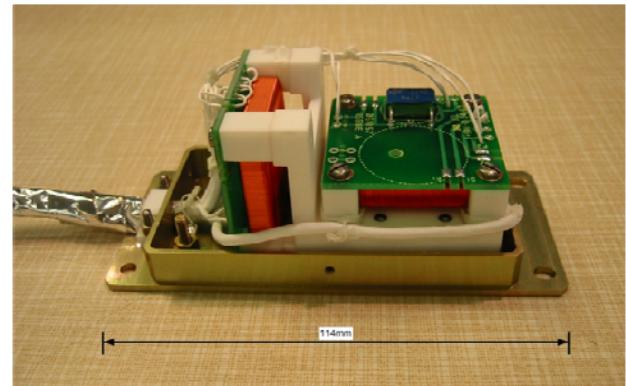
$$N_{\text{rms}} = \left( \int_{f_L}^{f_H} P(f) dt \right)^{1/2} = (P(1) \ln(f_H/f_L))^{1/2}$$

- Above Nyquist noise will be flat (ie white noise) due to ADC quantization
- Best quality fluxgates have NSD  $\sim 5 \text{ pT}/\text{Root Hz}$  at 1Hz

*Ripka (1998)*

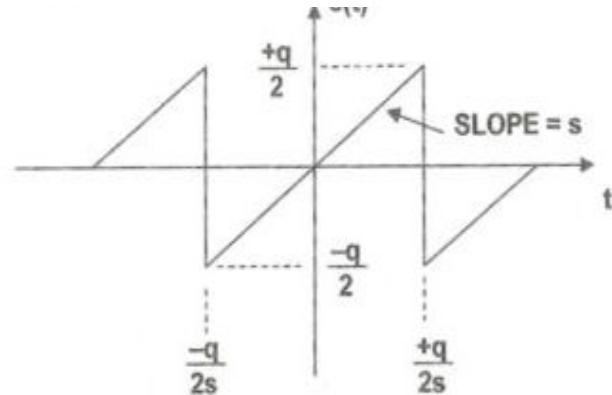
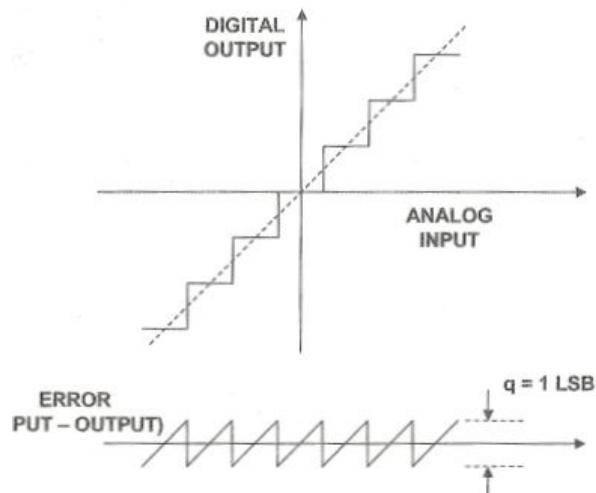
# Imperial fluxgate instrument performance

- Industrial partner - Ultra Electronics
- Cassini/Double Star Heritage
- Two core sensor
- Tuned second harmonic detection
- Dual sense and feedback windings
- Offset stability  $< 0.05 \text{ nT/}^\circ\text{C}$
- Scale factor drift  $< 40 \text{ ppm/}^\circ\text{C}$
- Noise density  $< 8 \text{ pT/root Hz } @ 1\text{Hz}$
- Operating range
  - $-80^\circ\text{C}$  to  $70^\circ\text{C}$  (operational)
  - $-130^\circ\text{C}$  to  $90^\circ\text{C}$  (non-operational)



# Importance of ADC: Quantization Noise

- Large number of bits  $N$
- Ideal linearity
- No missing codes
- Radiation tolerance
- Ideal quantization noise



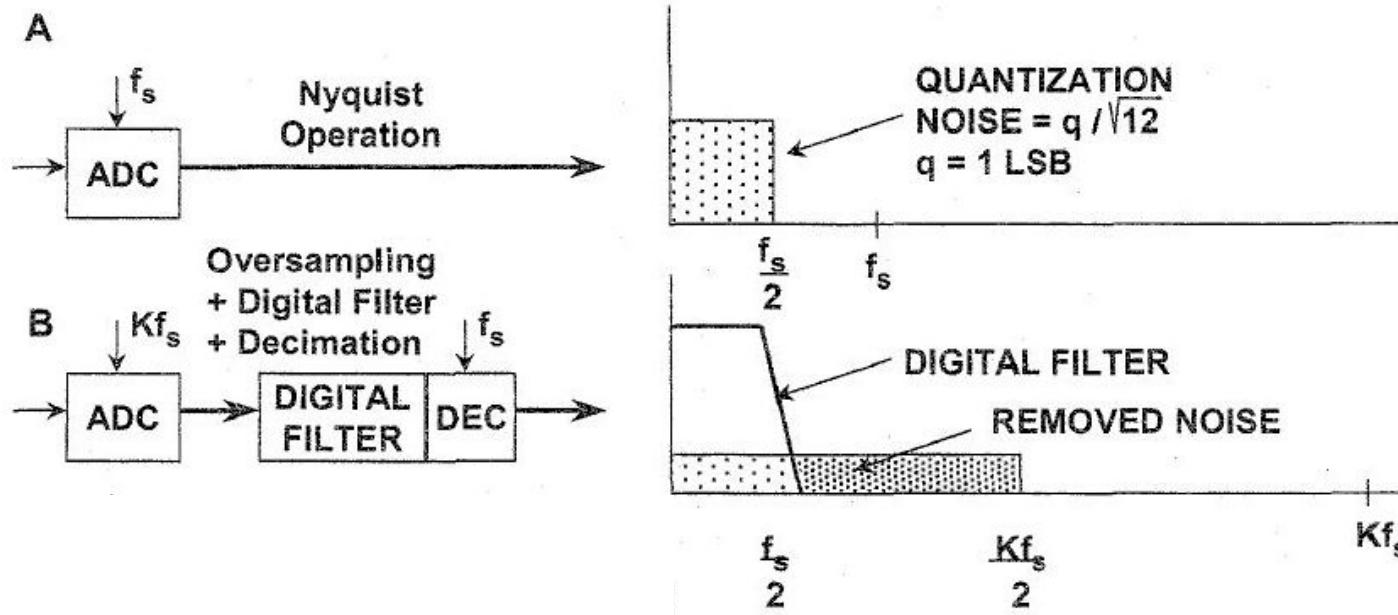
$$\diamond \text{ ERROR} = e(t) = st, \quad -\frac{q}{2s} < t < \frac{+q}{2s}$$

$$\diamond \text{ MEAN-SQUARE ERROR} = \overline{e^2(t)} = \frac{s}{q} \int_{-q/2s}^{+q/2s} (st)^2 dt = \frac{q^2}{12}$$

$$\diamond \text{ ROOT-MEAN-SQUARE ERROR} = \sqrt{\overline{e^2(t)}} = \frac{q}{\sqrt{12}}$$

# Quantization Noise

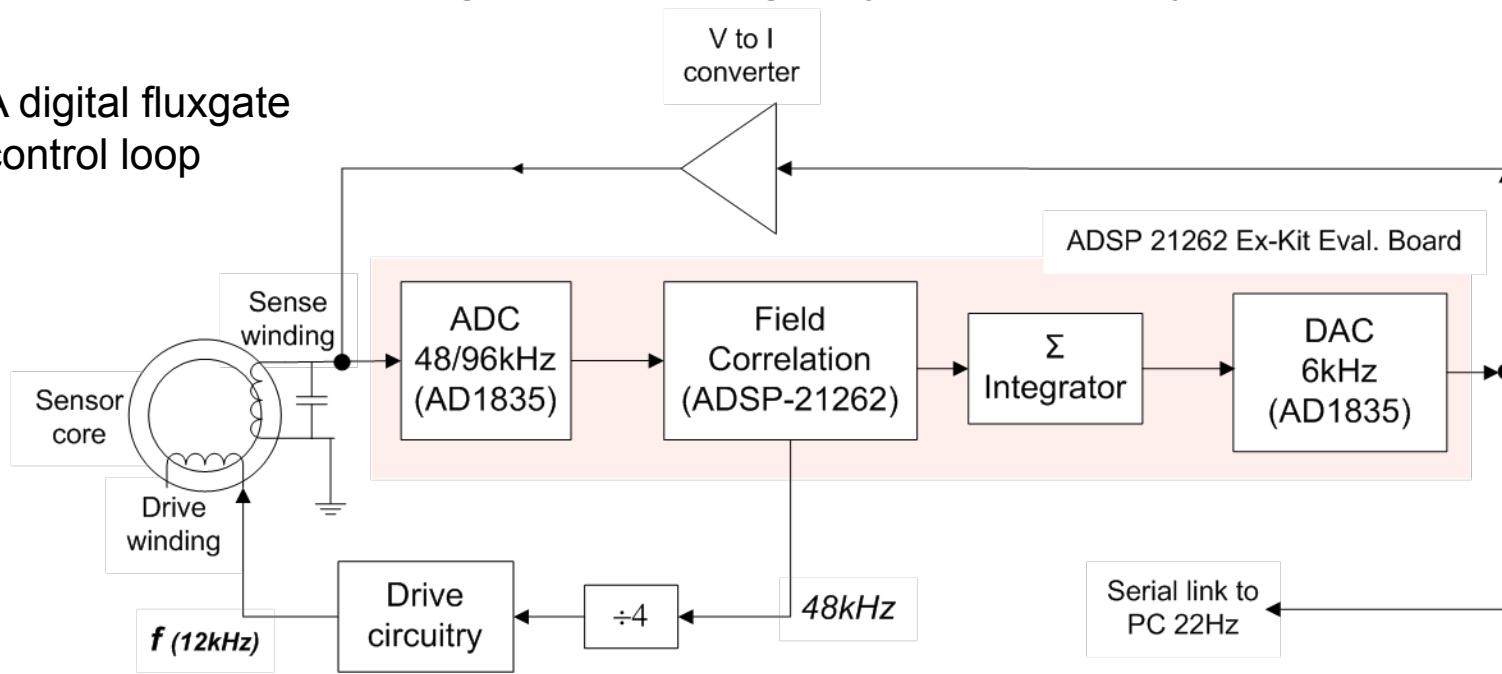
- Large N Rad-tolerant ADCs are a ‘big’ problem for all instruments
- Solution: MIL-STD devices with spot shields ( $N \sim 14$ )
- Traditionally a separate self contained card – Cluster, Rosetta, Cassini
- Use oversampling to reduce Q noise
- Q noise should be matched to intrinsic sensor noise based on desired range, resolution sensor scale factor and N and LSB



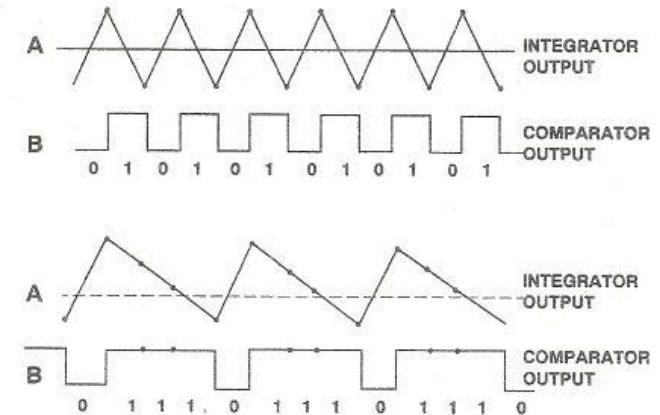
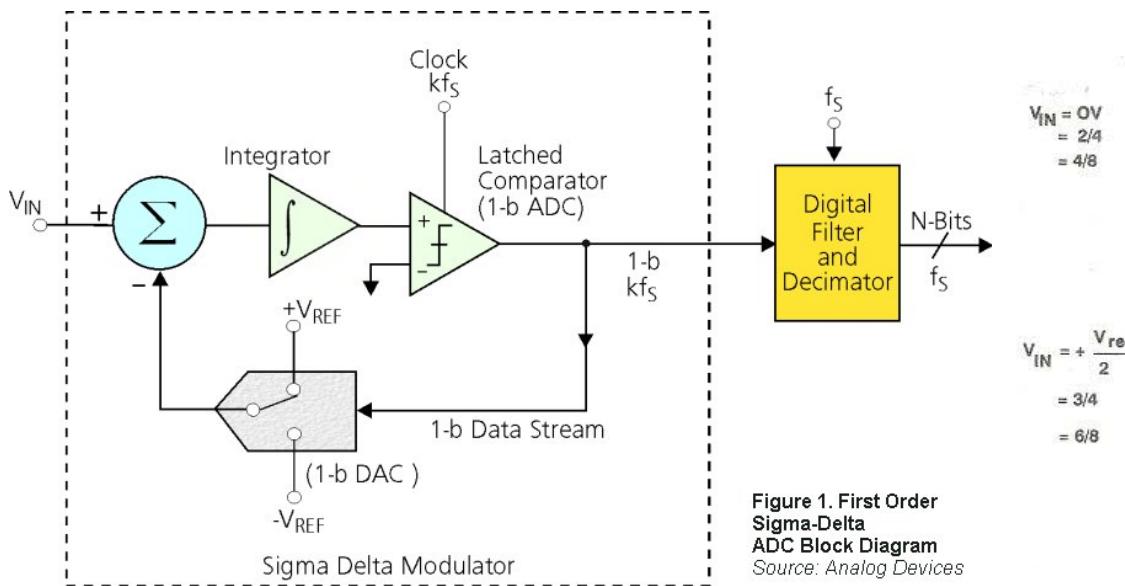
# Digital Magnetometers

- Means migrating control loop into digital domain
- ADC **and** DAC utilised **within** sensor control loop
- Offers increased flexibility - programmable
- First Missions late 90s - ROMAP, VEX, Astrid, Oersted
- Shown to reduce analogue content and power consumption
- Numerous designs – still being played out - a very active field

A digital fluxgate control loop

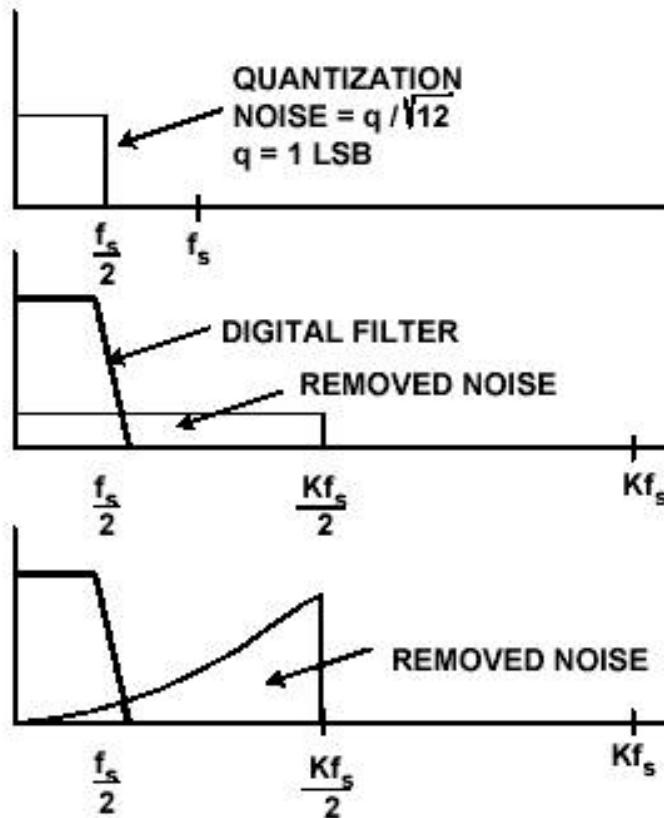
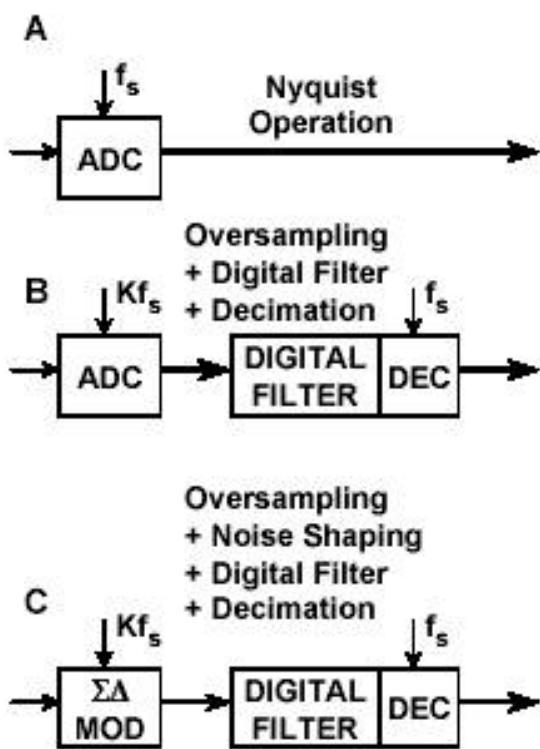


# Delta Sigma Fluxgates – A hot topic

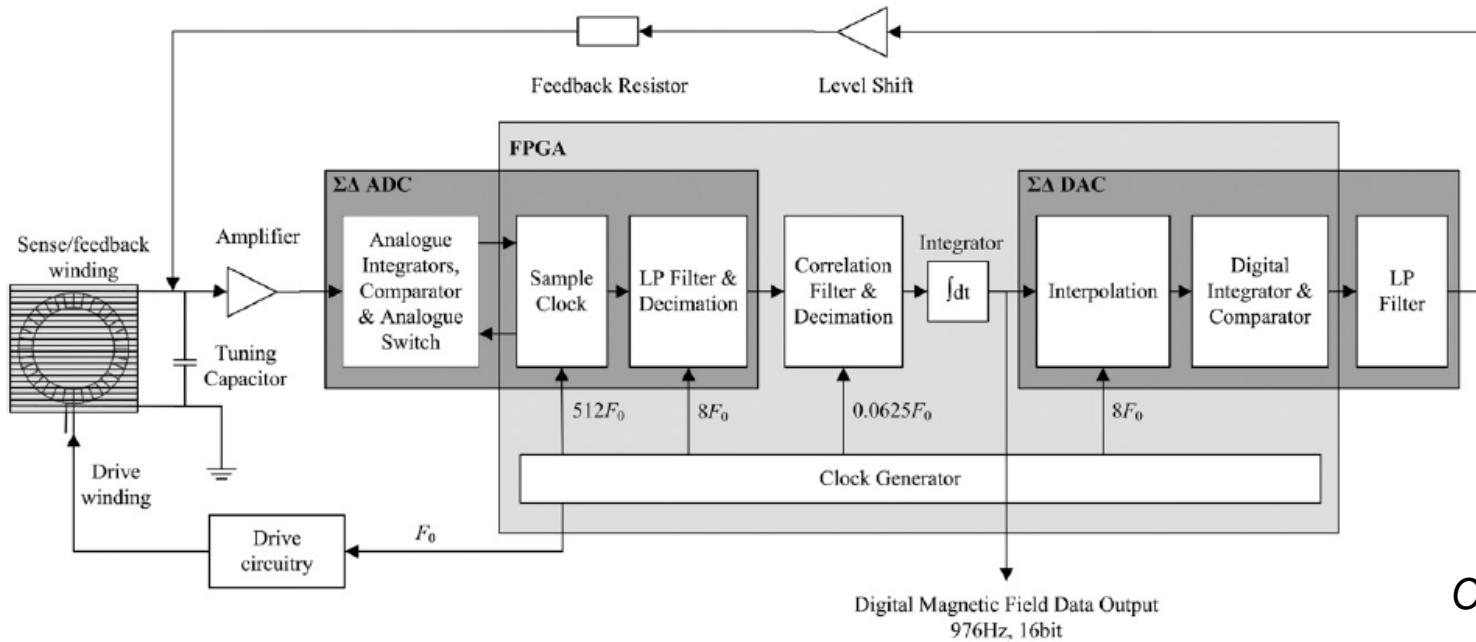


- Single bit quantization at very high frequency— linear by definition
- Tracks **changes** in consecutive samples rather than absolute value
- ‘Ones’ density of the 1 bit data stream provides an average value of  $V_{IN}$
- Can be implemented with a rad-hard analogue discrete and rad-hard digital logic – mixed signal ASIC
- Additional gain due to noise shaping
- Eliminates need for old fashioned non rad-hard ADCs

# Noise shaping effect

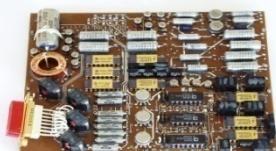


# Delta-Sigma Magnetometer



O'Brien (2007)

Replace



+



+

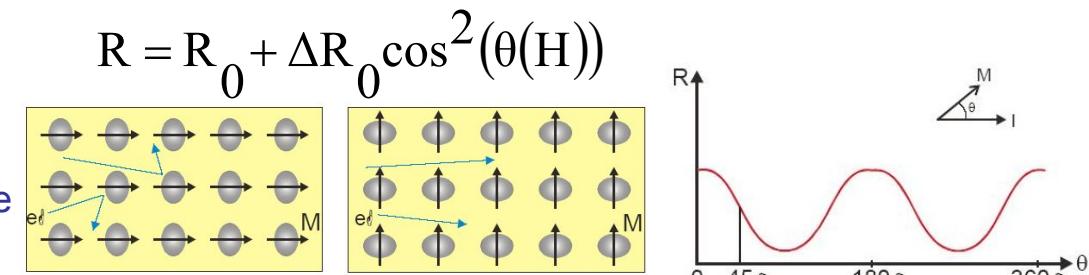


with

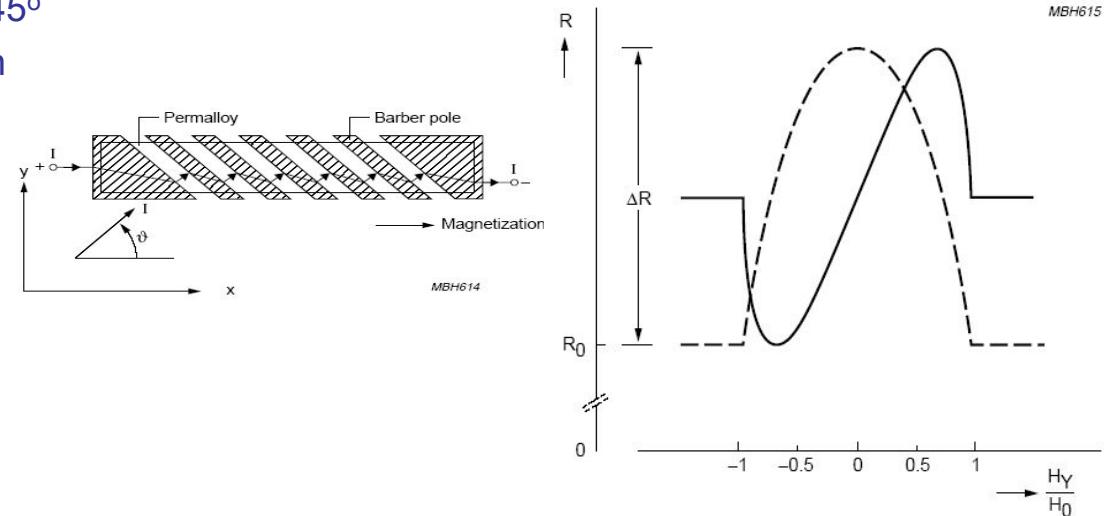


# Anisotropic Magnetoresistance

- **Magneto Resistance Effect**
  - Change of resistance in magnetic field
  - AMR single layer permalloy,
  - AMR  $\Delta R/R_{\min}$  of order 1- 2%
  - AMR has lowest noise floor
  - Johnson noise limited - no shot noise



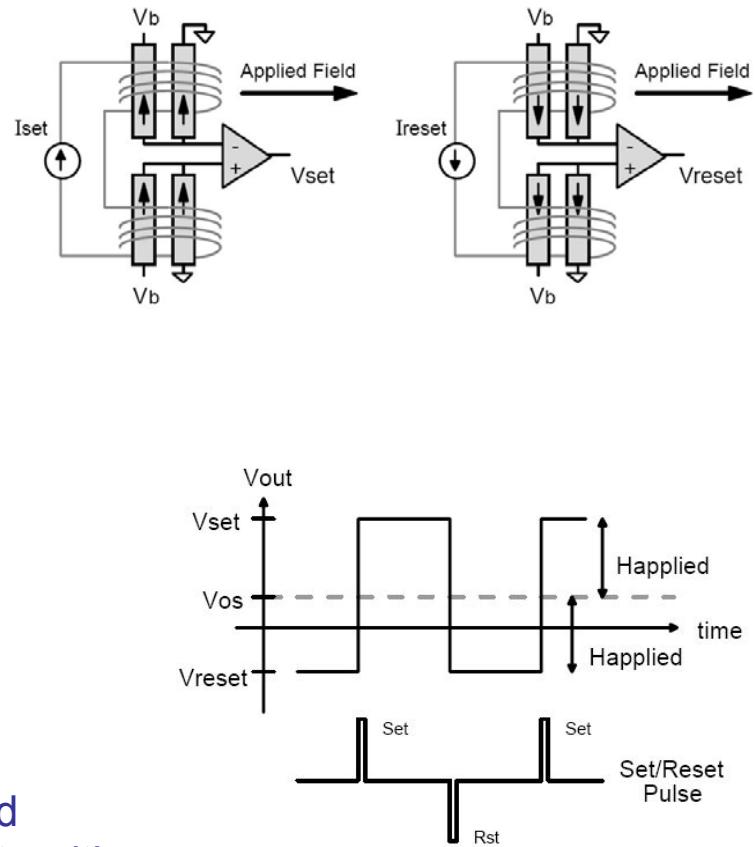
- **Barber Poles**
  - Max. sensitivity & linearity at  $M \vee H 45^\circ$
  - Conductive strips for linear operation



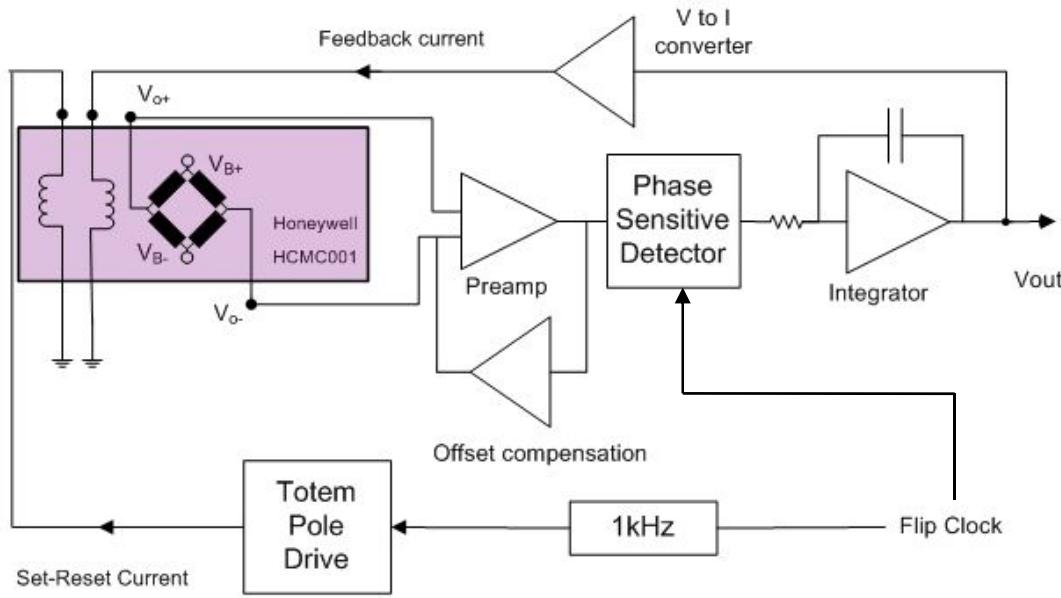
- **AMR Sensors**
  - Thin film solid state devices
  - Implemented as Wheatstone bridge
  - Mass <1g, Ceramic package
  - Sensitivity increases with increasing bridge voltage,  $V_B$

# Integrated coils

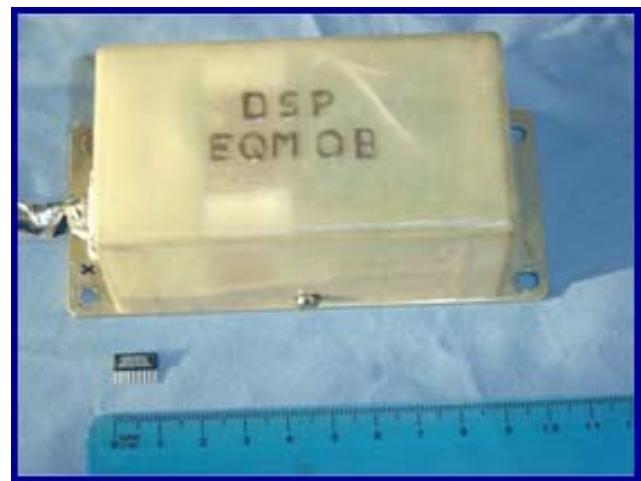
- **Set - Reset Coils**
  - Planar coils around each bridge resistor
  - Coil axis parallel to Easy axis
  - Used to re-align the anisotropic direction
  - Large current spike needed
  - Can extract sensor offset (unlike fluxgate)
  - Requires de-modulation to DC
  - Compensates for offset and offset drift
  - Improves sensor noise floor
- **Offset coils**
  - Integrated coils around the bridge
  - Coil axis parallel to Hard (sensitive) axis
  - Permits electromagnetic feedback
  - Used in closed loop back off measured field
  - Improves linearity and variation of sensitivity with temperature
  - Suppresses Barkhausen noise



# Single axis AMR magnetometer

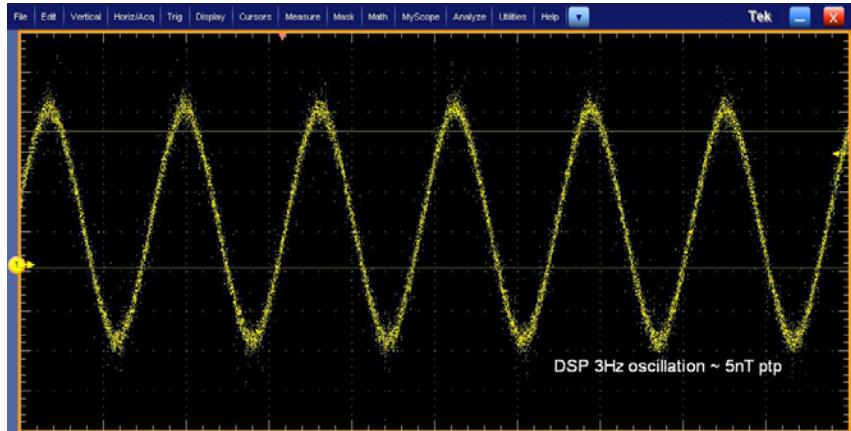


Analog build  
Set-Rest 4A with  $2\mu\text{s}$   $\tau_C$   
Sensitivity proportional to  $V_B$   
Closed loop



$$V_o = H_y \times \frac{R_{FB}}{A_{COIL}}$$

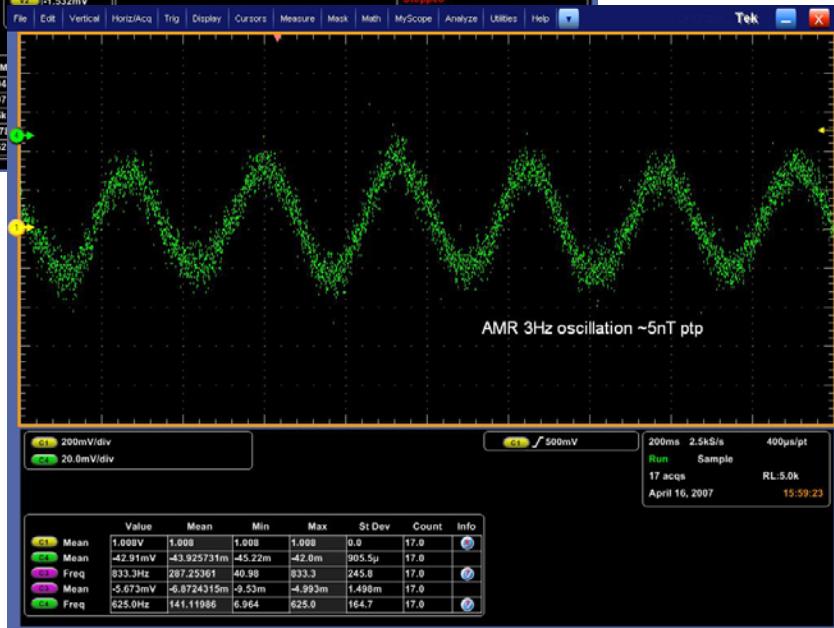
# Stimulus measurement – Fluxgate vs AMR



DSP (20mV/div scale)

- Three layer Mu-Metal shield
- 3Hz sine wave – 5nT ptp
- Optimal AMR configuration
- Closed loop,  $R_{FB}=9\text{k}\Omega$
- Bridge voltage 12V
- Offset compensation
- Flip frequency, 1.1kHz
- Sensitivity  $\sim 11\text{mV/nT}$
- Sensitivity not linear with increasing  $R_{FB}$
- Some residual offset in closed loop
- Temperature measurement outstanding

AMR  
(20mV/div  
scale)

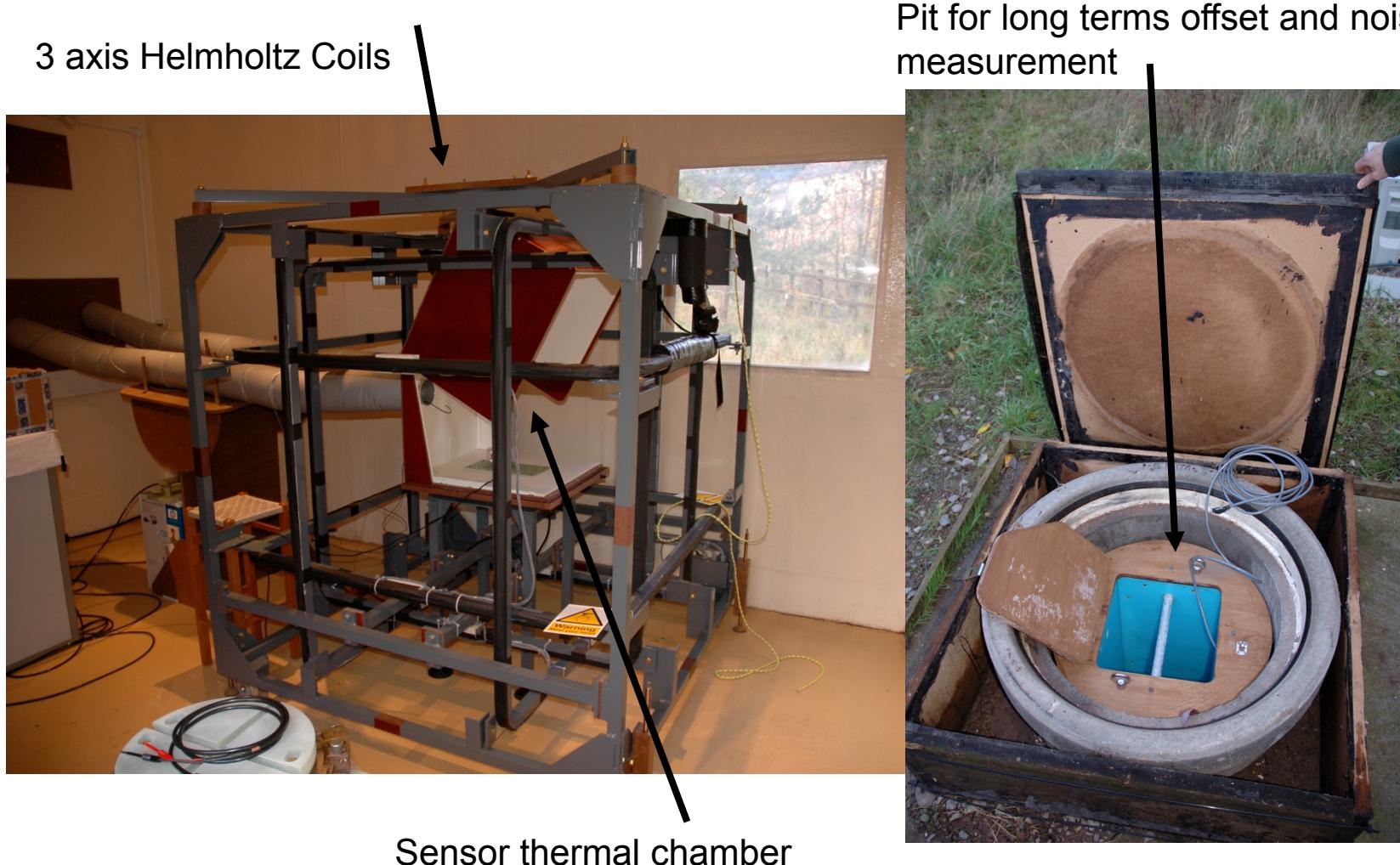


# Calibration equation for a vector magnetometer

- **Calibration Matrix** 12 parameters needed to transform measured volts to accurate field components into a physically useful co-ordinate system eg GSE, GSM
  - Calibration Matrix
    - Sensor **gains** – convert from raw volts to nT
    - Sensor **mis-alignments** – correct from deviation from nominal sensor axis
    - Euler **angles** – transform orthogonalised components into required system
  - Offset vector:
    - Sensor offset - correct for zero level readings (due to sensor, electronics or S/C)
  - Calibration Files
    - Text files with calibration matrix & offset vector for each sensor on a daily or orbit basis :

$$\begin{pmatrix} B_x \\ B_y \\ B_z \end{pmatrix} = \begin{pmatrix} c_{11} & c_{12} & c_{13} \\ c_{21} & c_{22} & c_{23} \\ c_{31} & c_{32} & c_{33} \end{pmatrix} \begin{pmatrix} B_{S1} - O_1 \\ B_{S2} - O_2 \\ B_{S3} - O_3 \end{pmatrix}$$

# Imperial's Magnetic Coil Facility



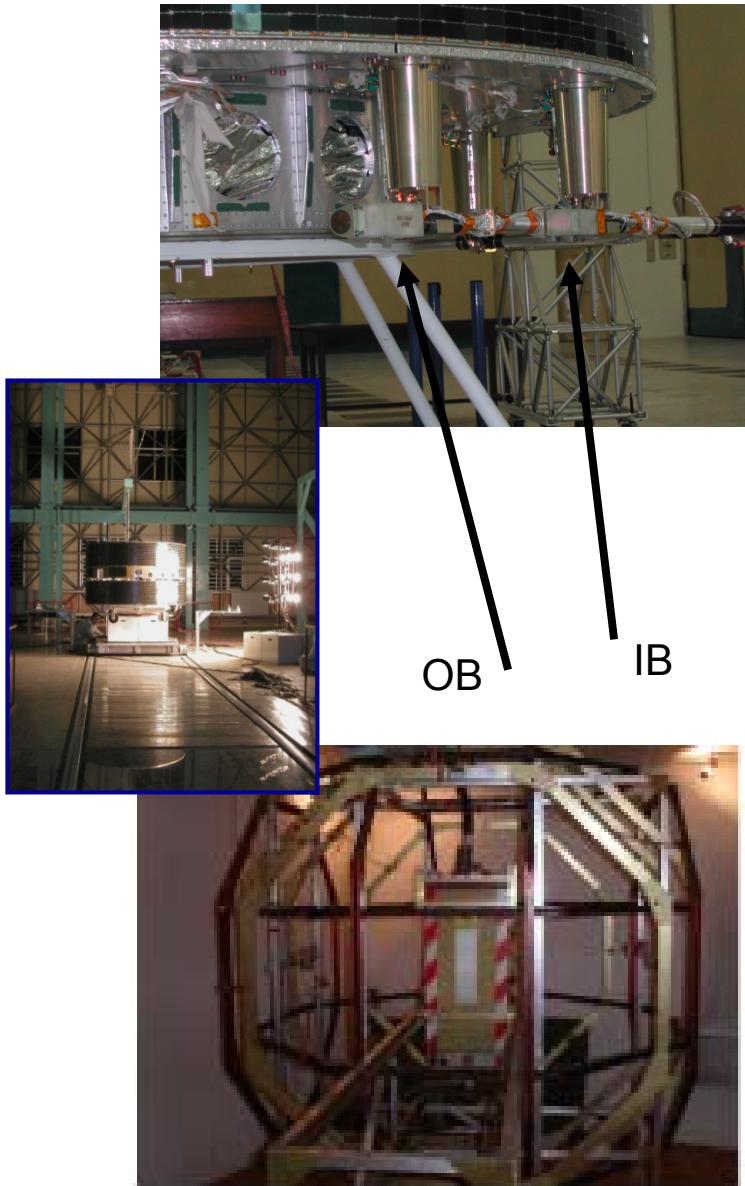
## Sensor under test



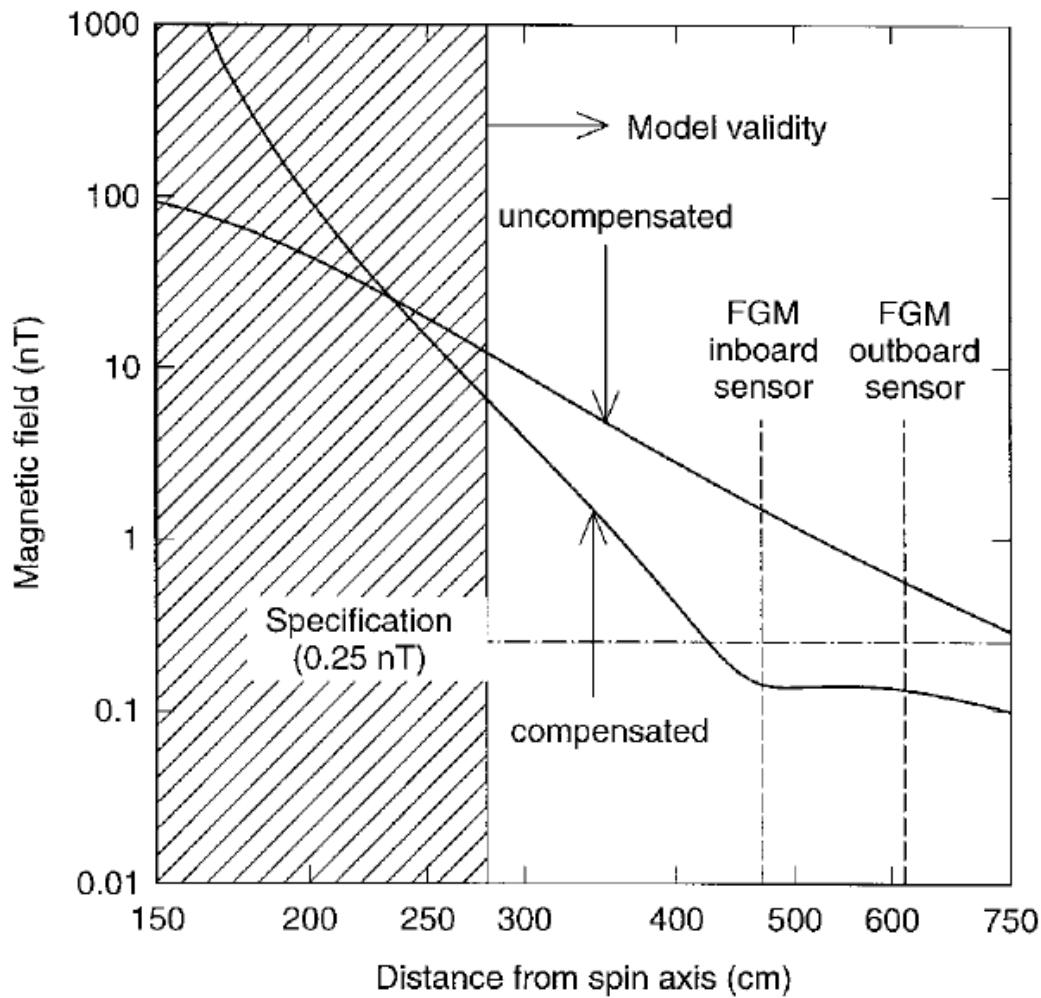
- Facility dynamically backs of Earth's field using
  - Two Earth Field Reference Magnetometers (EFR) located either side of the hut
- EFR located in pits either side of test hut
- Sum (average) of EFRs used to cancel Earths field inside coil system
- Difference (gradient) of EFRs used for monitoring

# Practical calibration models

- **Ground Calibration - we determine**
  - Sensor calibration parameters on ground,
  - Their associated temperature coefficients,
  - Their variations with input power
  - The sensor noise
- **Magnetic Cleanliness Program - includes**
  - Maximum length boom
  - Low field requirement at boom tip
  - Magnetic screening of materials and units
  - A spacecraft magnetic model
  - System level magnetic test
- **In-flight**
  - range switching, calibration steps
  - In-flight calibration techniques
  - Use of multiple sensors
  - Use of absolute and vector sensors
  - Use of dual-gradiometer modes



# System Level Magnetic Test: Cluster Example

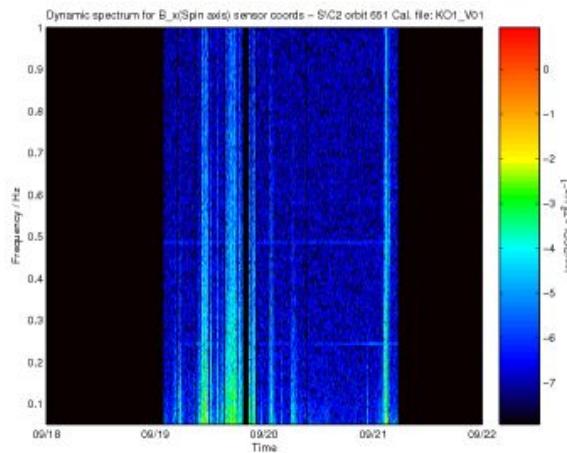
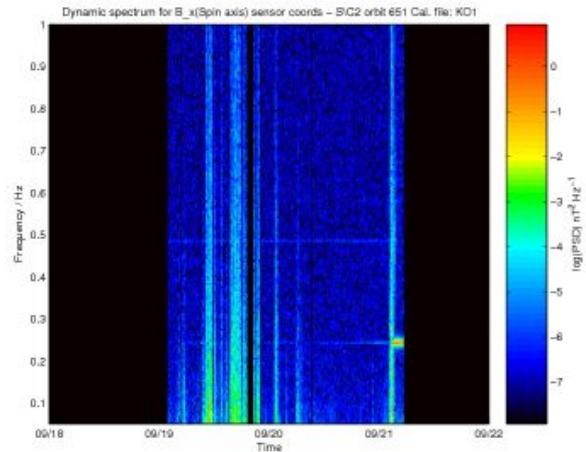


Cluster had a very rigorous (and expensive) magnetic cleanliness program

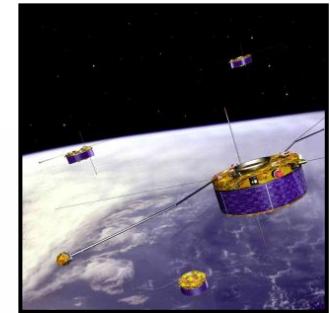
A S/C magnetic field of  $< 0.25\text{nT}$  is almost NOT the case on the vast majority of S/C

# In-flight calibration techniques

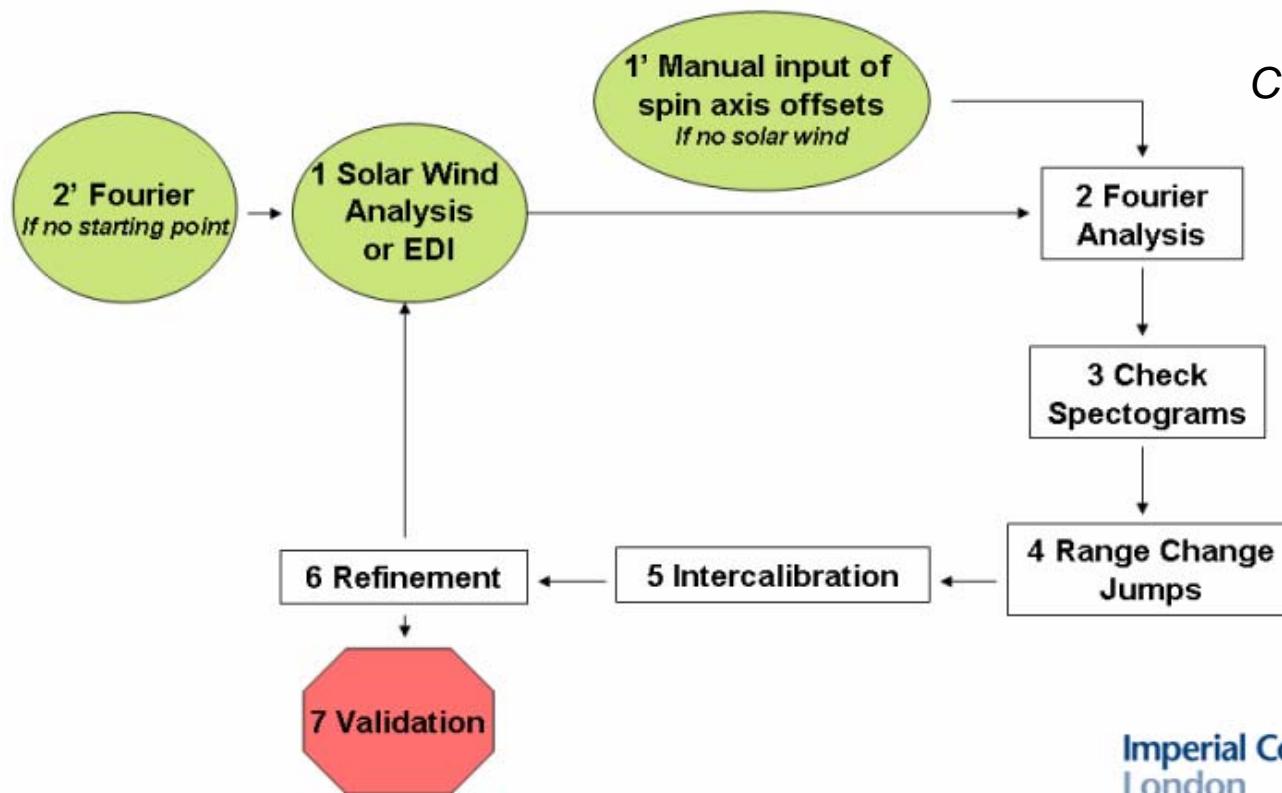
- **Spin stabalised spacecraft**
  - Fourier analysis on spinning data
  - Permits recovery of 8 of the 12 cal parameters
  - Major error – spin axis offset
  - Residual spin tone indicates calibration error
  - Example Missions:
    - Cluster, Ulysses, Double Star, Equator-S, Themis
- **Three-axis stabalised spacecraft**
  - More difficult to calibrate
  - Utilise S/C rolls for offset measurement
  - Statistical analyses of solar wind data
  - Looks for correlations between **B** and **B** components
  - Additional absolute reference magnetometer useful
  - Example Missions:
    - Cassini, Rosetta, Oersted, Venus Express
- **Multiple spacecraft missions = multiple calibration references**



# Multi-S/C Calibration – Cluster Example



## Cluster FGM Active Archive: Process Diagram



# Dual Magnetometer Mode

- Used in cases where S/C field contaminates measurement
- IB and OB sensor used as a gradiometer
- Ambient field same at both IB & OB
- S/C field NOT same at IB & OB
- Number of sensors is proportional to multipole moment that may be extracted
- Two sensors limit model to a dipole of fixed position
- Other techniques utilising pattern recognition in operation
- Relative sampling of both sensors important especially on spinning S/C
- Usually results in reduced data rates
- Example missions: Double Star, Venus Express

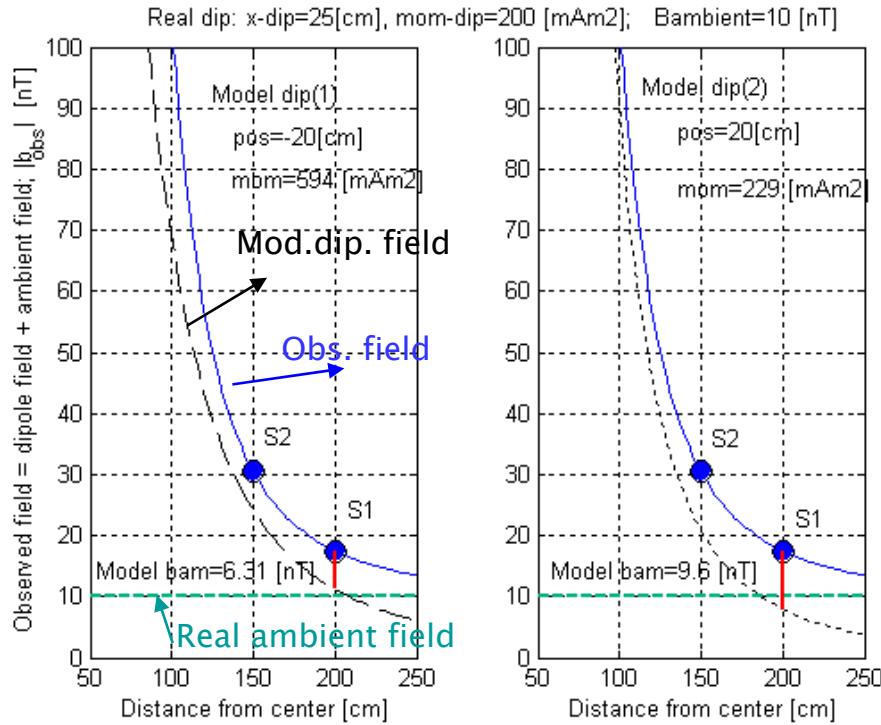
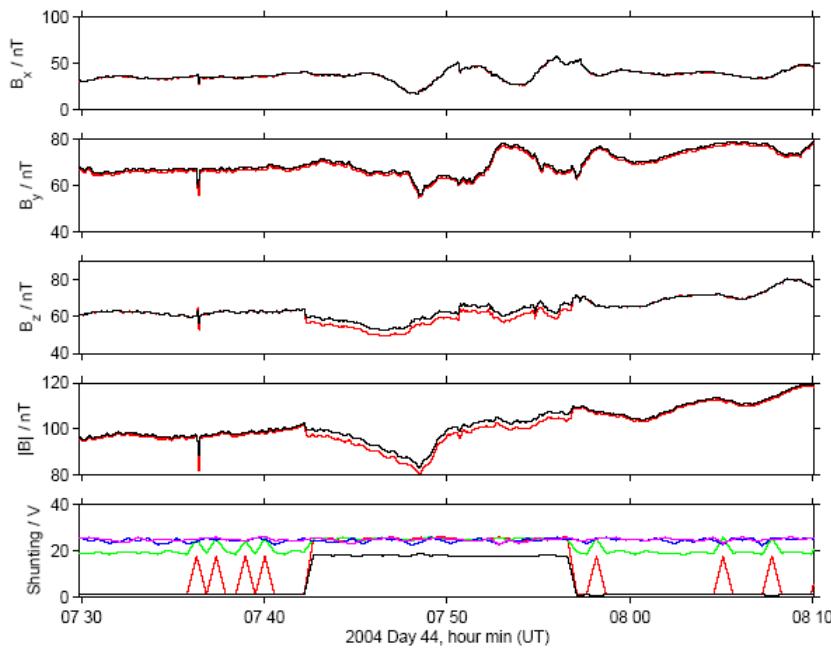


Figure courtesy Delva

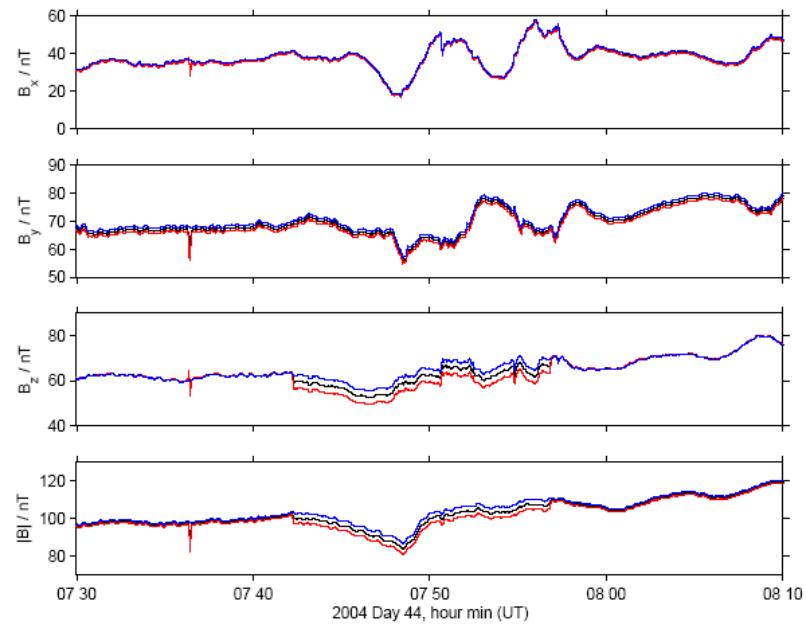
**Example (1 dim.):** **Solution 1:**  
 Real  $B_{\text{amb}}=10 \text{ nT}$   $p^{\sim} = -20 \text{ cm}$   
 Real SC dipole:  $m^{\sim}=594 \text{ mA}^2$   
 $p = 25 \text{ cm}$   $B_{\text{amb}}^{\sim}=6.31 \text{ nT}$   
 $m = 200 \text{ mAm}^2$  **Solution 2:**  
 $p^{\sim} = +20 \text{ cm}$   
 $m^{\sim}=229 \text{ mAm}^2$   
 $B_{\text{amb}}^{\sim}=9.6 \text{ nT}$

# Case Study. Double Star magnetometer

- OB sensor 5m, IB sensor 3.5m
- Spin synchronised disturbance due to unbalanced solar array current
- Amplitude varies with S/C shunting mode
- Data cleaned using gradiometer mode
- Resulting data set is spin averaged resolution (0.25Hz) compared to 11Hz on-board



Un-cleaned data and shunting modes

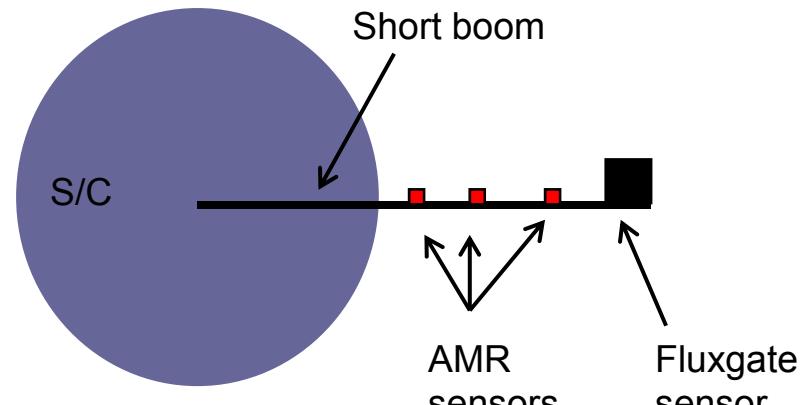


Un-cleaned and cleaned data

Carr (2005)

# A new magnetometer model?

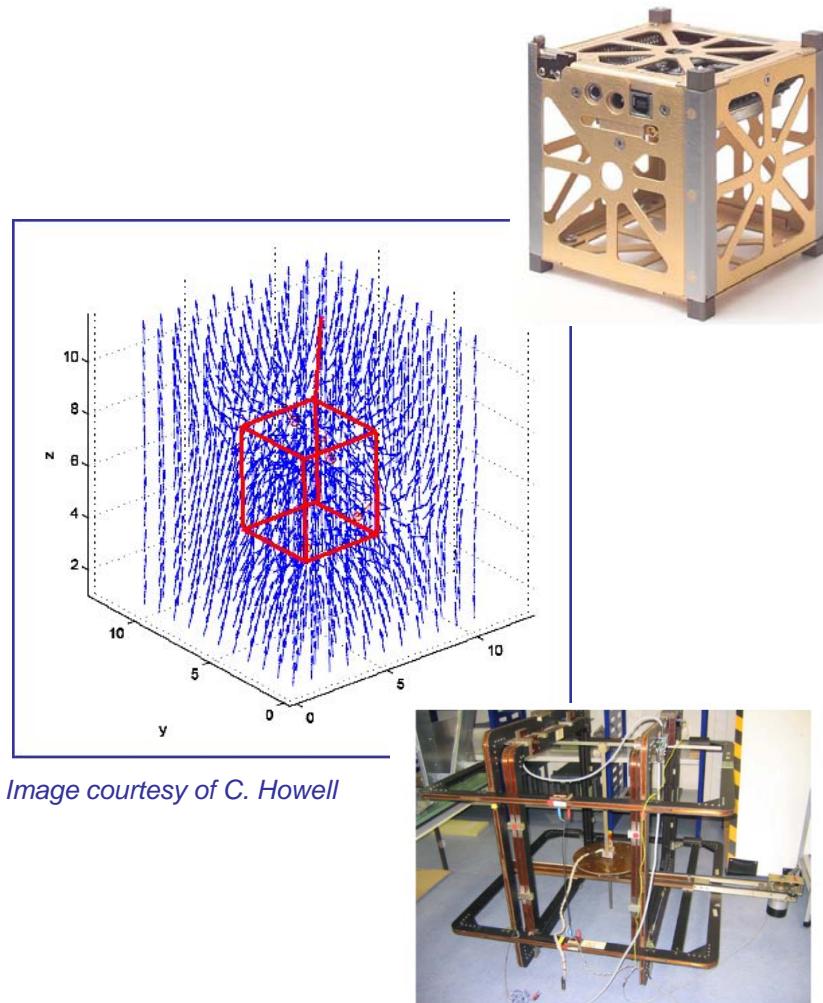
- Fluxgate - AMR combination
- Single fluxgate at end of a (shorter boom)
- Several AMR sensors inward of the fluxgate
- Permits multipole expansion of S/C field
- Accurate separation ambient field at instrument intrinsic data rate
- Precise tracking of fluxgate offsets
- Required for space plasma constellations
- Potential for automation
- Could be applied across missions
- Extendable to an array of AMR sensors



Question – How to validate concept ?

# A magnetometer array

- **Imperial College student satellite program**
  - Milestone - Two spacecraft in LEO
  - 10cm cube, 1kg modules
  - Injection into LEO approx \$30,000
- **Aims**
  - Measure ULF wave field in dayside magnetosphere
  - Flight qualify FPGA controlled AMR array
  - Validate S/C field rejection algorithms
  - Extract accurate magnetic field vector
- **Ground validation**
  - Mobile Coil Facility
  - ESTEC MDM to calculate E-box moments
  - Measure both S/C components and assembled S/C
  - Measured moments fitted to S/C model
  - Permits validation test of field rejection on ground

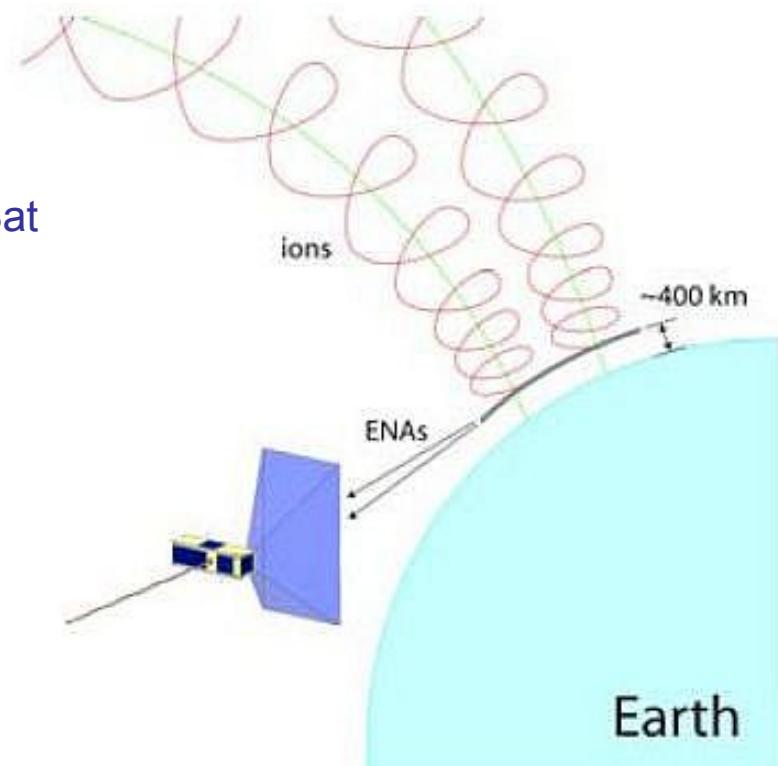




# Potential Flight Opportunity 2011

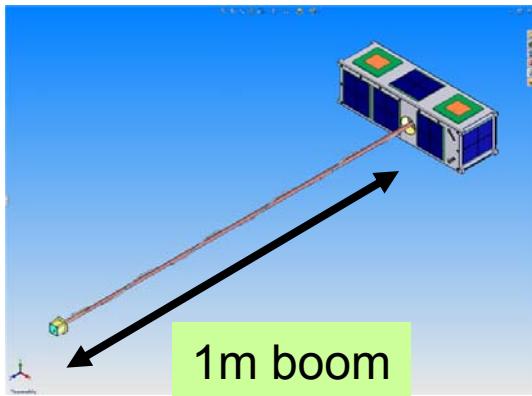
## TRIO CINEMA- CubeSat for measurement of ions, neutrals and magnetic fields

- Collaboration between UCB, IC & KHU
- Space plasma science measurement on 3U CubeSat
- Led by UCB/SSL
- Three 3U CubeSats
- LEO with  $>65^\circ$  inclination (72° nominal), 650km
- 1m deployable boom
- Spin stabilised at  $\sim 1\text{rpm}$
- Two MAG sensors
- Funder by NSF Space Weather Competition
- Heritage: GeneSat & STEREO

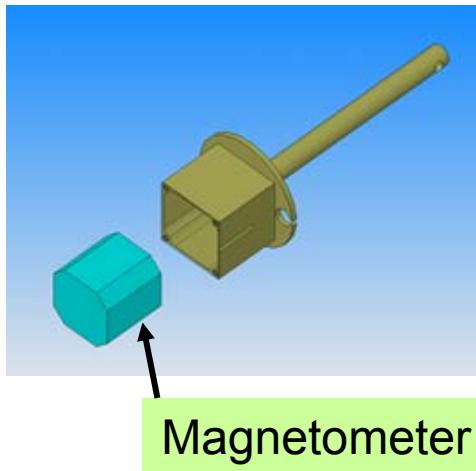




# CINEMA Magnetometer



1m boom



Magnetometer

1m extendable boom  
Boom mass  $\sim$ 120g  
MAG orientation not controlled  
Determined by magneto-torquer pulse post deployment  
Following de-tumble CubeSat spun up and spin axis aligned normal to ecliptic

## MAGIC Magnetometer Modes

### Attitude Mode

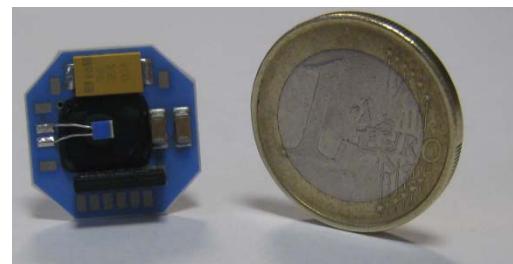
Accuracy  $<25\text{nT}$ ,  $<150\text{mW}$

### Science Mode

Accuracy  $<2\text{nT}$ ,  $<750\text{mW}$

Instrument Range  $+/-65536\text{nT}$

Resolution:  $0.25\text{nT}$



MAGIC Sensor head



MAG Boom Harness